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MERCURY DROPLET SIZE
AND DISTRIBUTION IN GLASS
CONDENSER TUBE IN 1-g AND
ZERO-GRAVITY ENVIRONMENTS

by Jonas I. Sturas
Lewis Research Center
Cleveland, Ohio

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By Jonas I. Sturas

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NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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MERCURY DROPLET SIZE AND DISTRIBUTION IN GLASS CONDENSER TUBE IN 1-g AND ZERO-GRAVITY ENVIRONMENTS

by Jonas I. Sturas

Lewis Research Center

SUMMARY

An experimental study was made of mercury droplet size and flux distribution in a horizontal convectively cooled glass mercury vapor condenser tube in 1-g and zero-gravity environments. The film analysis showed that the liquid phase was dispersed throughout the vapor phase in the form of discrete droplets, even in the higher vapor-quality range studied. The dispersed droplet breakup and coalescense in two-phase flow occurred simultaneously and continuously along the length of the condenser tube. The observed droplet size distribution was best described by the modified gamma distribution function.

Droplet size and flux distribution obtained from a series of high-speed film measurements, at various velocities and quality conditions along the tube length, were described statistically and fitted to a modified gamma distribution. The numerical values of three characteristic parameters that describe the gamma distribution curve varied with vapor velocity and quality.

The critical droplet size measured from a droplet track left on a condenser wall at its incipient movement and the maximum stable droplet size measured in a turbulent vapor stream were essentially linear functions of vapor quality and vapor velocity.

INTRODUCTION

The design and development of liquid-metal Rankine cycle electric power generating systems for space flight applications have defined a need for improved methods of predicting pressure drop and heat-transfer characteristics of two-phase flows associated with boiling and condensing processes in variable gravity environments. The condensing process is of particular interest since, in a Rankine turbogenerator system, the condensitions of the condensition of the con

ser component has stringent limitations on pressure level and pressure drop imposed by the turbine and circulating pump requirements.

The accuracy of analytically derived thermofluid dynamic relations is keyed to the assumption of realistic flow models that accurately characterize the formation, orientation, and transport of the liquid-phase portion of the two-phase flow. It can also be imagined that the flow characterization may be significantly affected by the imposed gravity environment and that the degree of this effect may be different at various flow conditions in a given tube geometry and locally in a given tube at a fixed flow condition.

Prior studies directed toward analytical definition of two-phase frictional pressure drop for mercury vapor condensation in nonwetting condenser tubes have been conducted by Koestal and Gido (ref. 1) and Heller (ref. 2) based on a postulated 1-g flow model.

As part of an overall program conducted at the NASA Lewis Research Center on mercury vapor condensing characteristics in 1-g and zero-gravity environments (ref. 3), a specific study was made to gain a better understanding of flow regimes in a typical condenser tube. The results discussed herein are based on high-speed motion-picture analysis of nonwetting mercury vapor condensing flow regimes in a uniformly cooled glass tube under 1-g and zero-gravity conditions. The analysis includes definition of droplet sizes and frequency distribution, maximum stable drop size, and critical drop size on the condenser tube wall at incipient movement. Empirical relations were derived from measured data for calculating droplet size distribution and other droplet parameters pertinent to two-phase-flow mechanisms.

Available on loan is a Lewis motion picture (C221) that includes an introduction to space power systems, a description of the MECA - AJ-2 flight package and associated hardware, and high-speed film sequences of mercury condensing in 1-g and zero-gravity environments. (A request card is included at the back of the report.)

Experimental data were obtained for a range of mercury weight flow rates from 0.315 to 0.435 pound per second, and the dependence of droplet distribution on vapor velocity and vapor quality was determined by photographic studies at various positions along the condensing length.

APPARATUS AND PROCEDURE

The experimental apparatus, system components, instrumentation, and test procedures were essentially the same for tests performed on the ground and in the zero-gravity flight facility.

System and Components

The test apparatus is shown schematically in figure 1, and the system flow diagram is shown in figure 2. The apparatus consisted of an expulsion tank that supplied triple-distilled mercury to the boiler. The boiler, consisting of a preheater and an electrically heated boiling section, produced a continuous supply of mercury vapor at a quality of 90 percent or greater. To assure consistent vapor flow and to minimize liquid carryover, the vapor was passed through a plenum chamber filled with stainless-steel cuttings. The mercury vapor flow was controlled to the desired value by a pressure regulator and a calibrated orifice.

The condensing tubes consisted of 0.440-inch constant-diameter low-alkali borosilicate glass tubing for the ground experiments and 0.400-inch constant-diameter transparent high-silica glass tubing for the aircraft flights. The overall length of the condenser tube in all cases was 87 inches. The tube was uniformly cooled by a crossflow of nitrogen gas issuing from two diametrically opposed manifolds, as shown in figure 1. The mercury vapor was condensed completely inside the tube, and the liquid mercury passed through an orifice at the end of the tube into the receiver. The desired condensing pressure was maintained by controlling the receiver cover gas pressure.

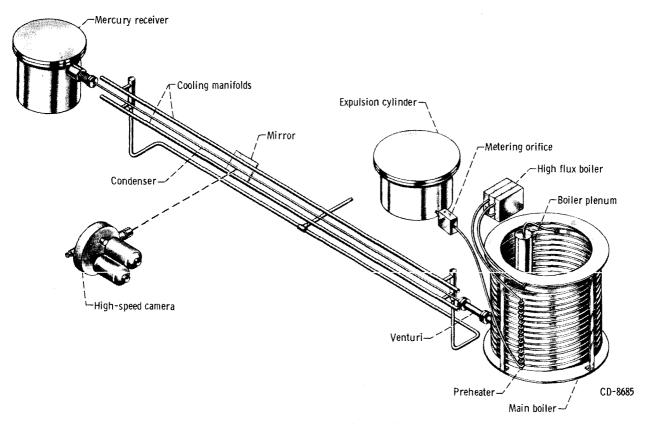
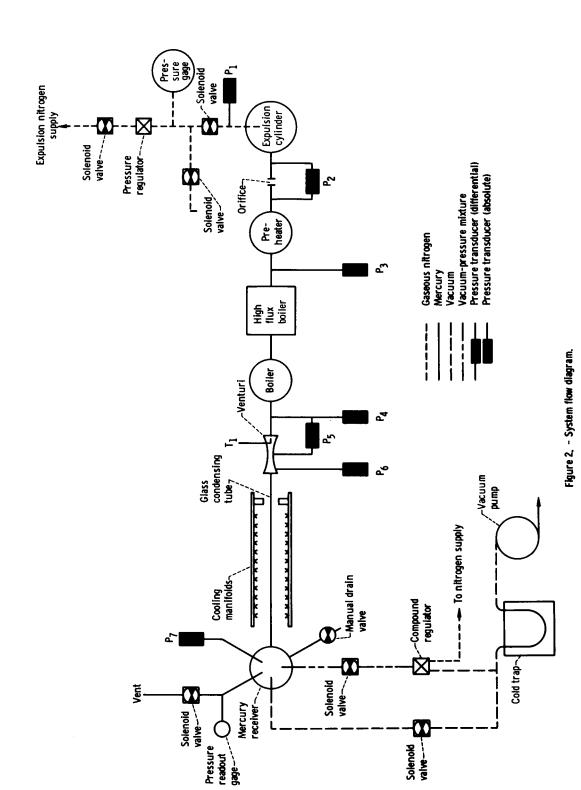


Figure 1. - Schematic drawing of test loop.



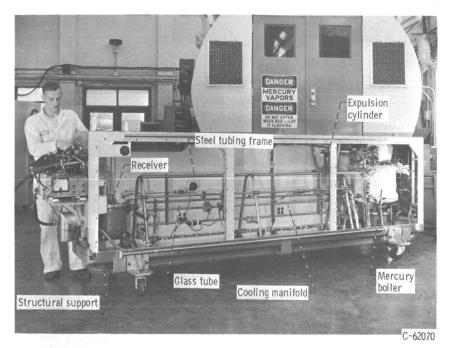


Figure 3. - Basic test system subassembly.

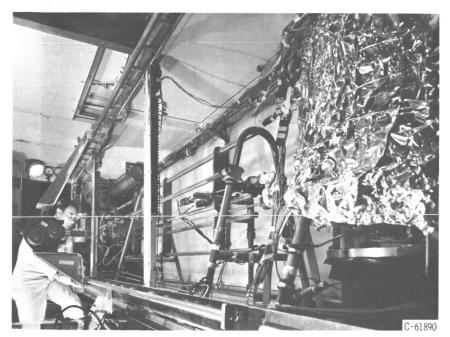


Figure 4. - Experimental system installed in bomb-bay of aircraft.

The experimental system installed in the flight package enclosure is shown in figure 3. The experimental system installed in the bomb-bay of the AJ-2 aircraft for the zero-gravity flights is shown in figure 4.

Films of the condensation process were taken with a 16 millimeter high-speed camera at rates ranging approximately from 1000 to 8000 frames per second. Although the initial camera-speed acceleration produced a nonuniform frame rate, the transients involved and the frame rate at any film position could be accurately determined from the curves shown in figure 5. These curves, one for each film taken, were obtained from the measurement of the number of frames between timing marks, which were recorded on the edge of the film in the form of light flashes at the rate of 120 impulses per second, as illustrated in figure 6.

In addition to the high-speed camera, a low-speed camera with a wide-angle lens was used in the aircraft to record the gross movement of the mercury interface during zero-gravity maneuvers. The vertical acceleration was indicated by an accelerometer that was in the field of view of the camera. The arrangement of the camera is shown in figure 7.

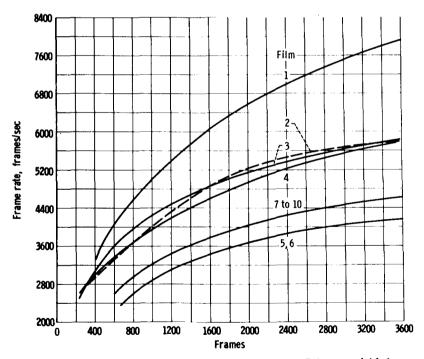


Figure 5. - Frame rate is function of number of frames. Reference point is from frame following last perforated number.

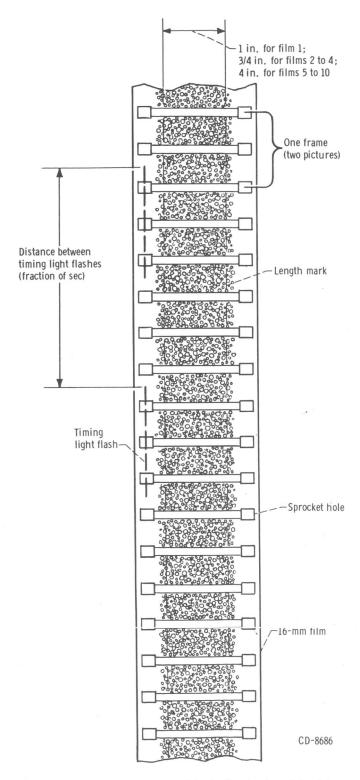


Figure 6. - Schematic drawing of 16-millimeter film showing relation between timing light flashes and frames recorded along film.

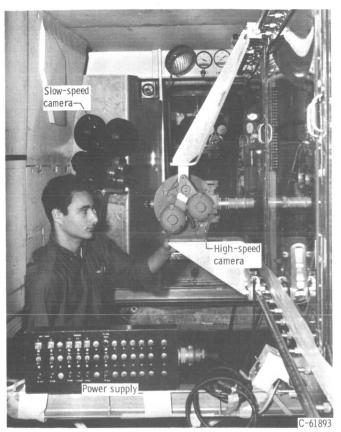


Figure 7. - Arrangement of camera relative to test rig used in airplane experiment.

Instrumentation

The mercury vapor flow rate into the condensing tube was measured by a venturi with a throat diameter of 0.277 inch and an exit diameter equal to the condenser tube diameter.

Oscillographs were used to record the data in both facilities. In the aircraft, the accelerations generated along the three axes (longitudinal, lateral, and vertical) were sensed by accelerometers located in the bombbay near the geometric center of the experiment. The "g" level indications were used for aircraft control throughout the maneuver and also were recorded to provide a direct correspondence with system pressures. The oscillograph trace provided a time history of the zero-gravity maneuver starting from pullup to pullout. About 4 or 5 seconds of the initial trajectory

following pullup were required to damp out pressure oscillations induced by the pullup maneuver. The portion of the trajectory during which the high-speed camera was operating was also recorded.

Stainless-steel inductance-type pressure transducers, capable of operating in a mercury environment up to 900° F, were used to measure venturi absolute pressure and venturi pressure drop. Low-temperature transducers were used at all other locations in the system. Each transducer in direct contact with mercury was oriented so that its mounting tube and core axis were on the same horizontal plane as the condensing tube to minimize the head effect. In addition, the mounting tube and core axis were positioned parallel to the lateral axis of the aircraft, which had the least acceleration. Oriented in this way, the g effect of the mercury column in the pressure line was minimal.

Temperatures throughout the system were measured by I. S. A. calibration K Chromel-Alumel thermocouples. A shielded sheathed thermocouple was immersed in the mercury vapor upstream of the venturi. At other locations, the thermocouple junction was spot welded to the outside surface of the various components such as the heaters.

Test Procedure

Prior to each test, the mercury loop was evacuated to approximately 0.060 torr, and the mercury heaters brought to operating temperatures. Mercury flow was initiated and maintained at a low rate for approximately 5 minutes keeping the liquid-vapor interface in the receiver to minimize the presence of noncondensables in the system. The receiver pressure was then increased to the operating value (between 14 and 15 psia), and the gaseous nitrogen flow for condenser tube cooling was regulated to locate the interface at the desired location. In the ground facility, the high-speed pictures were taken at several stations along the condensing length. In this way, condensing characteristics at these stations were recorded for a single setting of flow rate.

In the aircraft, data were taken at both 1-g and zero-gravity conditions. Data at 1 g (level flight) were determined by the stability of the aircraft in all axes. When the signal from the pilot was given on attaining this condition, the oscillograph recorders and high-speed camera were set in motion. To obtain a complete record of the zero-gravity period, the recording was begun immediately prior to the actual zero-gravity maneuver while still in level flight. The oscillograph recorders and the slow-speed wide-angle-lens camera were started and continued to operate throughout the entire maneuver (the aircraft and zero-gravity maneuver are discussed in ref. 3). During the maneuver, the interface was seen to undergo quite violent movements before stabilizing within the field of view of the high-speed camera. At this point of stabilization, when zero gravity was attained, the high-speed camera was set in motion and operated through the camera reel capacity.

RESULTS AND DISCUSSION

General Description of Film Observations

In order to provide an interpretive background for the quantitative study of droplet distribution in a condenser tube, it is helpful first to consider a general qualitative description of the photographic data.

The visual analysis on a film analyzer showed that the liquid droplets in two-phase mercury vapor flow in 1-g and zero-gravity environments appeared dispersed throughout the vapor phase approximately in the proportion shown in table I.

The shape of the droplets ranged from spheroids to cigar shaped. The following four shapes were the most frequent on the analyzer screen:

- (1) Spherical small droplets moving in the vapor stream
- (2) Spheroid larger than the most frequent droplet size; moves close to the condenser wall with no oscillation (termed maximum stable droplet)

TABLE I. - PROPORTION OF DROPLETS DISPERSED IN VAPOR PHASE

Film	Percentage of	f all droplets	Assumed	Environ-	
	Moving on or near condenser wall	Moving in core of vapor stream	linear vapor quality, x	ment, g's	
		Ground test	_		
1	50	50	0.82	1	
2	75	25	. 36	1	
3	80	20	. 20	1	
4	85	15	. 03	1	
		Airplane test			
5	70	30	0.03	0	
6	60	40	. 03	0	
7	50	50	. 20	0	
8	70	30	. 03	1	
9	80	20	. 03	1	
10	70	30	. 20	1	

- (3) Ellipsoidal moves along the bottom on the condenser wall
- (4) Irregular unstable droplets in oscillatory motion on the bottom of the condenser wall close to the vapor-liquid interface

The ratio of the droplet velocity in the vapor core to that of droplet velocity on the condenser wall varied approximately from 5, close to the inlet, to 2, near the vapor-liquid interface. All the droplets originated on the condenser wall and remained stationery until additional condensation increased the size to the "critical diameter", at which time the vapor drag started to slide them along the wall. The droplet population in a condenser tube increased very rapidly with a decrease in vapor quality. An increase in the droplet population resulted mainly from the breakup process of coalesced droplets.

The effect of gravity on droplet distribution across the condenser tube was pronounced. In zero gravity, droplets were somewhat smaller and more uniformly dispersed across the condenser tube than in a 1-g environment. This effect is discussed in more detail in the following section.

Analysis

Based on the general qualitative description of droplet dispersion throughout the vapor phase, observed from the films taken of the condensation process, a model was postulated to account quantitatively for a droplet size distribution along the condenser tube. The analysis was carried out in two steps: The first step was to seek the distribution of drop-

let frequency, droplet density, and droplet flux along the length of the condenser tube. The second step was to obtain the diameter of a droplet, originated by condensation on the condenser tube wall, at its incipient movement resulting from vapor drag. This diameter is termed the "critical droplet size." In this step, the height of the most frequently occurring droplet on the condenser wall was also obtained.

<u>Basic model</u>. - The basic model of droplet dispersion in the condenser tube is based on the following considerations:

- (1) All droplets originated by condensation on the condenser tube wall. The droplets remained on the wall until reaching a critical diameter after which they started to slide toward the interface.
- (2) Sliding droplets along the condenser wall increased in size due to coalescence. In most instances, the large droplets were torn apart, and were entrained in the vapor stream as a result of turbulence.
 - (3) Droplet coalescence and breakup were random in nature.
- (4) The droplets were dispersed nonuniformly across the condenser tube. The droplet population on the condenser wall and in the core of the vapor stream varied with vapor quality and gravity conditions.
- (5) The droplets were assumed to have a gamma distribution. This assumption is based on a marked similarity between the shape of the histograms of droplet size distribution obtained from measurements and the shape obtained with the gamma probability density function.
- (6) As reported by Eckert (ref. 4), in fully developed turbulent flow in a pipe, the laminar-sublayer thickness is well represented by the following relation:

$$\delta_{\rm b} = 63.5 \, \rm d(Re_{\rm v})^{-0.875}$$

The laminar-sublayer thickness was obtained from this relation δ_b for ground test conditions, and the results are shown in table II. (All symbols are defined in appendix A.)

TABLE II. - LAMINAR SUBLAYER THICKNESS FOR
GROUND TEST CONDITIONS

Film	Vapor quality, x	Vapor Reynolds number, Re _v -0.875	Laminar-sublayer thickness, ^δ b' in.
1	0. 82	13. 52×10 ⁻⁵ 28. 50×10 ⁻⁵ 37. 60×10 ⁻⁵ 2. 27×10 ⁻³	0.004
2	. 36		.008
3	. 20		.010
4	. 03		.063

Therefore, in the range of the condensing length investigated, the ratio of the measured droplet height δ_y of the most frequently occurring size droplet on the condenser wall to the calculated thickness of the laminar sublayer and the ratio of the thickness of the laminar sublayer to the height (taken as $\delta_c/2$) of the critical droplet size varied, respectively, as follows:

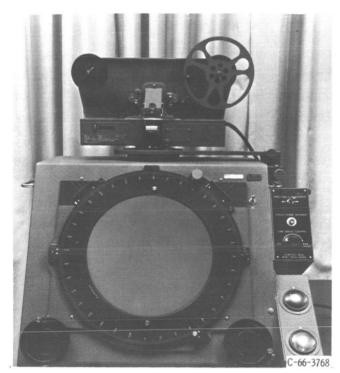


Figure 8. - Motion-picture analyzer.

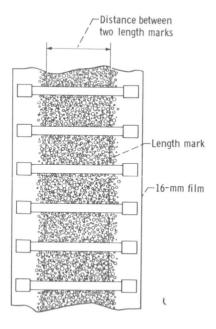


Figure 9. - Schematic representation of dro; indistribution as shown on analyzed screen.

$$2.75 \le \frac{\delta_y}{\delta_b} \le 0.23$$

$$1.54 \le \frac{\delta_{\mathbf{b}}}{\delta_{\mathbf{c}/2}} \le 1.05$$

From these results, the following two assumptions were postulated:

- (1) With the exception of the droplets very near the liquid-vapor interface, the height of the most frequent size droplet occurring on a condenser wall would protrude above the laminarsublayer thickness and would be in the region of the turbulent layer.
- (2) All the droplets on the condenser wall will be originated by a process of condensation within the laminar-sublayer thickness.

Droplet size distribution. - The droplet sizes and population were developed with the aid of the motion-picture analyzer shown in figure 8. Depending on the lens used, the magnification from film to the analyzer screen was ×16.5 and $\times 27.0$ for the ground tests, and $\times 1.4$ and $\times 2.2$ for the airplane tests. The measurements and counts were made of all droplets that appeared between the two length marks on the tube (as shown in fig. 9) in focus on the film analyzer screen. The measurement of the projected image of a droplet on the analyzer screen was accomplished with crosshair micrometer dials in the xand y-directions. The measurements of droplet sizes and counts of the numbers of each size were made at four specific locations (by two observers)

*along the film. For example, for film 1, the measurement of droplet sizes and a count of their number were made at frames 2000, 2200, 2500, and 3000. The frames were counted from the reference point indicated on each film.

The arithmetic mean of the ith interval of a droplet size analyzed was determined from the following relation

where

 $\delta_{z,i}$ midvalue of i^{th} interval, termed "group mark"

m number of intervals

f, number of droplets in ith group interval, called absolute group-mark frequency

δ, variable droplet diameter

N total (average) number of droplets counted between two length marks

The data, thus obtained at various vapor velocities and quality conditions, are listed in tables III(a) to (d) for ground tests (1-g environment) and in tables III(e) to (j) for airplane tests (1-g and zero-gravity environments). In order to compare the droplets of various sizes flowing at various velocities through the section considered on the same time basis, the absolute frequency of each group mark was related to 1 second. Thus,

$$n_{\mathbf{w}, i} = \frac{f_i}{\Delta \tau_{\mathbf{w}, i}} \tag{2}$$

the percentage of droplets on the wall, and

$$n_{s,i} = \frac{f_i}{\Delta \tau_{s,i}}$$
 (3)

the percentage of droplets in the stream of the vapor core. The total number of droplets

TABLE III. - MEASURED

(a) Film 1 (6 in. from inlet); within interval of frames 2000 to 3000

Meas-		Fr	ame		Meas-	Average	Group	Abso-	Time for	Time for
ured	2000	2200	2500	3000	ured	group	mark,	l	droplets	droplets of
drop- let	Nur	Number of droplets		fre- quency,	range, δ _z ,	δ _{z,i} ,	group- mark	of i th group	i th group mark to	
diam-					f,	7	in.	fre-	mark to	travel be-
eter,					number	in.		quency,	travel be-	tween two
δ _z ,					of drop-			f _i ,	tween two	length
in.					lets		}	number	length	marks in
								of drop-	marks of	stream of
								lets	condenser	vapor
									wall,	core,
									$(\Delta \tau)_{w,i}$	$(\Delta \tau)_{s,i}$
									sec	sec
0.005	1	0	2	2	1]					
. 007	4	2	5	0	3 }	0.005 to 0.007	0.007	4	0.01862	0.01295
. 008	3	1	6	2	3)			:		
. 009	5	7	3	4	5	0.008 to 0.009	. 009	8	. 02620	. 0197
		•	•	_						
. 010	4	3	5	3	4 }	0.010 to 0.011	. 011	11	. 03110	. 01860
. 011	7	8	6	8	7 }					
. 012	3	0	2	4	2]	0.012 to 0.013	.013	7	. 01742	. 00824
. 013	4	4	6	5	5 }	0.012 (0 0.013	.013	'	. 01 /42	. 00024
. 014	3	4	2	2	3]					
. 015	2	1	3	3	2	0.014 to 0.015	.015	5	. 01810	. 0082
. 016	1	0	1	2		İ		ì		
	-		_	_	1 }	0.016 to 0.017	.017	3	. 01470	. 00560
. 017	2	1	3	2	2 }			Ì		
.018	2	1	0	1	1]	0.018 to 0.019	. 019	2	. 01380	. 00486
. 019	0	2	1	1	1 }	0.010 (0 0.019	.019	4	.01300	.00100
. 020	1	0	1	0	0]					
. 021	1	2	0	1	1	0.020 to 0.021	. 021	1	. 01118	. 00435
Total				-				41		

DROPLET DISTRIBUTION

counted from reference point; ground test; 1-g environment

f _i δ _{z,i} , droplets sec	Percentage of droplets on wall, n _{w,i} , droplets sec	Percentage of droplets in vapor core, ns, i' droplets sec	Total droplets passing through reference point along con- denser tube, nt, i = nw, i + ns, i, droplets sec	Average number of drop-lets per unit volume of condenser tube, n _{V, i} , droplets in. 3	Average droplet flux, $\varphi_i = \frac{n_t}{A}, \frac{\text{droplets}}{(\sec)(\text{in.}^2)}$	n _{t,i} δ _{z,i}	Total area of histo- gram, A _{t,i}
0.028	107	154	261	35	1730	1.83	0. 0875
. 072	153	203	356	70	2340	3.21	. 1201
. 121	177	296	473	96	3110	5.21	. 1580
. 091	201	424	625	61	4110	8.13	. 2090
. 075	138	305	443	44	3910	6.64	. 1480
. 051	102	268	370	26	2430	6.28	. 1240
. 038	72	206	278	17	1828	5.28	. 0927
. 021	45	115	160	9	1051	3. 36	. 0534
0.497		<u> </u>	2966	358		39.94	0.9927

TABLE III. - Continued. MEASURED.

(b) Film 2 (42 in. from inlet); within interval of frames 1000 to 2500

Measured		Fra			Measured	Average	Group	Absolute group-mark	Time for droplets of i th group mark to travel	Time for droplets of i th group mark to travel
droplet diameter, õ _z , in,			2000 f drop		frequency, f, number of droplets	group range, δ_z , in.	mark, δ _{z, i} , in.	frequency, f _i , number of	between two length marks of condenser wall, $(\Delta \tau)_{\mathbf{W}, \mathbf{i}}$,	
								droplets	sec	sec
0. 003 . 005	0	2	1 0	1 3	1 }	0. 003 to 0. 005	0. 004	2	0. 02461	0. 01315
. 006 . 007	2	1 0	2 2	2	2 }	0. 006 to 0. 007	. 006	4	. 02640	. 01134
. 008 . 00 9	4 2	3	7	5	5 }	0, 008 to 0, 009	. 008	6	. 01873	. 01290
. 010 . 011	5 2	5 2	4	5 2	5 }	0. 010 to 0. 011	. 010	7	. 02760	. 01635
. 012 . 013	5 3	5 2	5 2	4 3	5 }	0. 012 to 0. 013	. 012	8	. 014510	. 011350
. 014 . 015	5 4	6 4	6 5	6	6 }	0. 014 to 0. 015	. 014	10	. 01381	. 01137
. 016 . 017	7 5	8 5	7 6	7	7 5	0. 016 to 0. 017	. 016	12	. 01443	. 00877
. 018 . 019	5 3	7	5 4	6 2	6 3	0. 018 to 0. 019	. 018	9	. 011650	. 00672
. 020 . 021	6 2	5 4	4	7 5	5 }	0. 020 to 0. 021	. 020	8	. 01354	. 00616
. 022 . 023	1 2	5 0	7 4	9	5 }	0. 022 to 0. 023	. 022	7	. 01253	. 00693
. 024 . 025	6	2 2	4	5 0	4 }	0. 024 to 0. 025	. 024	6	. 01223	. 02780
. 026 . 027	5 2	2	0 2	4 3	$\left[\begin{array}{c} 3\\2 \end{array}\right]$	0. 026 to 0. 027	. 026	5	. 00926	. 00974
. 028 . 029	0	3	2	4 2	2 }	0. 028 to 0. 029	. 028	4	. 00970	
. 030 . 031	2	0	3	3	1	0. 030 to 0. 031	. 030	3	. 01990	. 01530
. 032 . 03 3	3	0	0	1	1 }	0. 032 to 0. 033	. 032	2	. 01950	. 01380
. 034 . 035	0	0	1 1	1	1 }	0. 034 to 0. 035	. 034	2	. 03060	
. 036	1	0	0	1	$\left[\begin{array}{c} 1\\ 0\end{array}\right]$	0. 036 to 0. 03	7 . 036	1	. 03120	
. 037	0	1		1			l			1

DROPLET DISTRIBUTION

counted from reference point; ground test; 1-g environment

fiōz, i,	Percentage of	_	Total droplets passing	Average number of	Average drop-	n _{t, i} δ _{z, i}	Total area o
droplets	droplets on	droplets in	through reference point	droplets per unit vol-	let flux,		histogram,
sec	wall,	vapor core,	along condenser tube,	ume of condenser tube,	$\varphi_{\mathbf{i}} = \frac{\mathbf{n_t}}{\mathbf{A}},$		A _{t, i}
	n _{w, i} ,	n _{s, i} ,	$n_{t, i} = n_{w, i} + n_{s, i}$	n _{v, i} ,			
	droplets	droplets	droplets	droplets	droplets		
	sec	sec	sec	in.3	(sec)(in, ²)		
0. 008	61	38	99	13	651	0. 040	0. 01368
. 024	113	86	201	26	1322	1. 20	. 02778
. 048	240	116	356	39	2342	2, 85	. 04920
. 070	190	107	297	46	1954	2.97	. 04105
. 096	413	176	589	53	3875	7. 06	. 08140
. 140	543	220	763	66 50		10. 76	. 10545
. 192	2 622 342		964	79	6342	15.40	. 13322
. 162	580	337	917	59	6033	16. 50	. 12673
. 160	442	325	767 53		5046	15. 38	. 10600
. 154	418	252	670	46	4408	14. 75	. 09259
. 144	367	53	420	39	2763	10. 10	. 05804
. 130	404	128	532	33	3500	13. 80	. 07352
. 112	310	0	310	26	2040	8. 66	. 04284
. 090	112	48	160	20	1053	4. 80	. 02211
. 064	77	36	113	13	743	3. 62	. 01562
. 068	49	0	49	13	322	1.67	. 00677
. 036	25	0	25	7	164	. 90	. 00346
1.698			7232	631		130. 82	0.99476

TABLE III. - Continued. MEASURED

(c) Film 3 (54.6 in. from inlet); within interval of frames 1200 to 2000

Measured			ame		Measured	Average	Group	Absolute	Time for droplets of	Time for droplets of
droplet diameter, δ_z , in.	$\delta_{\mathbf{z}}$, Number of droplets		Ь	frequency, group f, range, number of δ_z , droplets in.		mark, ⁸ z, i' in.	group-mark frequency, f _i , number of droplets	i^{th} group mark to travel between two length marks of condenser wall $(\Delta \tau)_{W, i}$ sec	i^{th} group mark to travel between two length marks in stream of vapor core, $(\Delta \tau)_{\mathbf{S}, i}$ sec	
0.004	1 3	0	2	1 5	2 }	0.004 to 0.005	0.005	6	0.03640	0.01320
. 006	4	2	3	3						
.007	6	3	7	5	3 } 5	0.006 to 0.007	. 007	8	. 02230	. 00860
. 008	0	2	1	2	1]					
. 009	7	7	8	9	8	0.008 to 0.010	. 009	9	. 01740	. 00885
. 010	4	6	5	5	5]					
. 012	7	5	4	3	5 }	0.010 to 0.012	. 011	10	. 01680	. 00746
. 013	15	9	14	13	13 }	0.013 to 0.014	. 013	21	. 02870	. 01410
. 014	8	7	10	6	8 ∫				1020.0	
. 015	11	9	12	7	10 }	0.015 to 0.016	. 015	18	. 02621	. 00710
. 016	7	8	11	5	8 J					
.017	8	7	8	6	⁷ }	0.017 to 0.018	. 017	13	. 01820	. 00765
.018	3	5	7	4	5 ∫ 2 }					
.019	7	6 4	5 6	7	6 5	0.019 to 0.020	. 019	11	. 01580	. 01360
. 020	8	5	4	7	6)					
.022	4	2	3	3	3	0.021 to 0.022	. 021	9	. 01970	. 00809
. 023	6	3	4	2	4]					
. 024	3	5	6	3	4 }	0.023 to 0.024	. 023	8	. 02460	. 01107
. 025	5	4	3	6	4 }	0.005 / 0.000		_		24.250
. 026	2	1	3	2	2 }	0.025 to 0.026	. 025	6	. 02470	. 01850
. 027	4	2	4	5	4 \	0. 027 t ₀ 0. 028	. 027	5	00100	. 02190
. 028	0	1	1	2	1 ∫	0.021 10 0.028	.021	3	. 03160	. 02190
. 029	3	2	1	3	2 }	0.029 to 0.031	. 029	3	. 02851	. 01392
. 031	1	0	1	1	1 ∫	0.020 to 0.031	. 020	J	. 02031	.01382
. 032	2	0	0	1	1 }	0.032 to 0.034	. 033	2	. 02873	. 01395
. 034	1	1	1	0	1)		l			
Total								129		

DROPLET DISTRIBUTION

counted from reference point; ground test; 1-g environment

$f_i^{\delta}_{z,i}$	Percentage of	Percentage of	Total droplets passing	Average number of	Average drop-	$n_{t,i}\delta_{z,i}$	Total area of
droplets	droplets on	droplets in	through reference point	droplets per unit vol-	let flux,	, -,-	histogram,
sec	wall,	vapor core,	along condenser tube,	ume of condenser tube,	n _t	1	A _{t,i}
	n _{w,i} ,	n _{s,i} ,	$n_{t,i} = n_{w,i} + n_{s,i}$	n _{v.i} ,	$\varphi_{\mathbf{i}} = \frac{\mathbf{n_t}}{\mathbf{A}},$		-,-
	droplets	droplets	$n_{t,i} = n_{w,i} + n_{s,i},$ droplets	droplets	droplets		
	sec	sec	sec	$\frac{\substack{n_{v,i},\\ \text{droplets}}}{\text{in.}^3}$	$(sec)(in.^2)$		
0.030	132	91	223	39	1467	1. 1150	0.0308
. 056	287	186	473	53	3112	3. 3110	. 0653
. 081	414	201	615	59	4047	5. 5350	. 0849
. 110	482	268	750	66	4935	8. 2500	. 1035
. 273	585	300	885 138		5823	11. 5050	. 1221
. 270	550	507	1057	118	6955	15. 8550	. 1459
. 221	570	341	911	85	5994	15. 4870	. 1257
. 2 09	556	162	718	72	4724	13.6420	. 0991
. 189	364	223	587	59	3862	12. 3270	. 0810
. 184	261	145	406	53	2671	9. 3380	. 0560
, 150	194	65	259	39	1704	6.4750	. 0357
. 135	126	46	172	33	1132	4.6440	. 0237
. 087	63	43	106	20	698	3.0740	. 0146
. 066	56	20	85	13	559	2. 8050	. 0059
2. 063			7247	847		113.363	0. 9941

TABLE III. - Continued, MEASURED.

(d) Film 4 (67.8 in. from inlet); within interval of frames 1000 to

Measured droplet			ame		Measured frequency,	Average group	Group mark,	Absolute group-mark	Time for droplets of i th group mark to travel	Time for droplets of i th group mark to travel
diameter, δ _z , in.	\vdash	Ь.	2000 f drop	2500 olets	f, number of droplets	range, δ_Z , in.	δ _{Z, i} , in.	frequency, f _i , number of droplets	between two length marks of condenser wall, $(\Delta \tau)_{W, i}$, sec	between two length mark in stream of vapor core $(\Delta \tau)_{8,1}$, sec
0. 00 9 . 010	3 4	1 3	2 5	3 4	2 4	0. 009 to 0. 010	0. 010	6	0. 01034	0. 00364
. 011 . 013	5 6	2	6 5	3 4	5	0. 011 to 0. 013	. 012	9	. 01018	. 00359
. 014 . 015	7	9 7	6 6	10 6	8 }	0. 014 to 0. 015	. 014	14	. 01192	. 00632
. 016 . 017	14 5	15 7	14 9	17 8	15 7	0. 015 to 0. 017	. 016	22	. 01842	. 00653
. 018 . 019	17 9	15 11	16 10	19 10	17 }	0. 018 to 0. 019	. 018	27	. 01837	. 00642
. 020 . 021	16 10	12 14	15 8	11 15	14 }	0. 020 to 0. 021	. 020	23	. 01448	. 00842
. 022 . 023	12 8	15 4	10 6	14 5	13 6	0. 022 to 0. 023	. 022	16	. 01404	. 00376
. 024 . 025	9 6	8 10	10 7	8 8	9 }	0. 024 to 0. 025	. 024	15	. 01259	. 00631
. 026 . 027	7	11 5	9	12 3	10 }	0. 026 to 0. 027	. 026	13	. 01208	. 00672
. 028 . 029	6 5	4 7	8 4	10 4	⁷ ₅ }	0. 028 to 0. 029	. 028	12	. 01374	. 00686
. 030 . 031	7	4 2	6	6 2	6 }	0. 030 to 0. 031	. 030	9	. 01364	. 00756
. 032 . 033	3	4 3	5 2	3 6	4 }	0. 032 to 0. 033	. 032	8	. 01810	. 01210
. 034 . 035	6	6	4 2	5	5 }	0. 034 to 0. 035	. 034	7	. 02120	. 01760
. 036 . 037	2	3 2	2	4 3	$\left\{\begin{array}{c} 3\\2 \end{array}\right\}$	0. 036 to 0. 037	. 036	5	. 02790	. 01740
. 038	0	1 1	1 2	2	1)	0. 038 to 0. 039	. 038	3	. 01920	. 01850
. 040 . 042	1	1 0	0 2	2 1	1 }	0. 040 to 0. 041	.040	2	. 02290	. 01270
Total	L	L	L			I	1	191		I

, DROPLET DISTRIBUTION

2500 counted from reference point; ground test; 1-g environment

f _i δ _{z, i} , droplets sec	Percentage of droplets on wall, nw,i' droplets sec	Percentage of droplets in vapor core, ns, i, droplets sec	Total droplets passing through reference point along condenser tube, nt, i = nw, i + ns, i, droplets sec	Average number of droplets per unit volume of condenser tube, n _V , i, droplets in. 3	Average drop- let flux, $\varphi_i = \frac{n_t}{A},$ $\frac{\text{droplets}}{(\text{sec})(\text{in}.^2)}$	n _{t, i} ō _{z, i}	Total area of histogram, A _t , i
0.060	493	247	740	39	4 880	7.40	0.04887
. 108	747	373	1 120	59	7 360	13.42	. 07399
. 196	998	332	1 330	92	8 740	18. 61	. 08786
. 352	1014	506	1 520	145	9 980	24.30	. 10042
. 486	1125	355	1 480	178	11 750	26. 70	. 09777
. 460	1350	409	1 752	151	11 570	35. 20	. 11620
. 352	967	637	1 604	105	10 560	35. 30	. 10597
. 360	1012	357	1 369	99	6 980	33.60	. 09044
. 338	914	290	1 204	86	7 930	31, 30	. 07954
. 336	744	262	1 006	79	6 610	28. 20	. 06646
. 270	562	178	740	59	4 860	22. 20	. 04883
. 256	375	99	474	53	3 120	15. 20	. 03131
. 238	281	60	341	46	2 240	11.60	. 02253
.180	152	43	195	33	1 282	7. 02	. 01288
. 114	113	24	157	20	1 032	5. 97	. 01037
. 080	74	24	98	13	644	3.93	. 00647
4. 186			15 137	1257		328. 55	0. 9995

(e) Film 8 (47 in. from inlet); within intervals of frames 3200 to 3800 counted from reference point; airplane mean vapor velocity, 43.50 feet per second.

Measured droplet		Fra			Measured frequency,	Average group	Group mark,	Absolute group- mark frequency,	Time for drop- lets of i th group
diameter, δ_z , in.	3200 Nu	3400	3600 f drople	3800 ets	f, range, $\delta_{z,i}$, number of δ_z , in. droplets in.		f _i , number of droplets	mark to travel between two length marks of condenser wall, $(\Delta \tau)_{\rm W,i}$, sec	
0.013	1	0 2	0	1 1	0 }	0.013 to 0.014	0.014	1	
. 014	1 1 3	0	0	1	0 }	0.015 to 0.016	. 016	1	0. 02000
.017	0	1 0	3	1 2	1 2	0.017 to 0.018	. 018	3	. 02083
. 019 . 020	0 2	3 4	1 2	1 3	1 }	0.019 to 0.020	. 020	4	. 01156
. 021 . 022	4 2	0	5 4	2 2	3 }	0.021 to 0.022	. 022	5	. 02033
. 023 . 024	1 6	2 5	4 2	0 4	2 }	0.023 to 0.024	. 024	6	. 01408
. 025 . 026	3 9	1 7	7 8	4 5	7	0.025 to 0.026	. 026	11	. 02273
. 027 . 028	0 7	3	3 4	2 3	2 }	0.027 to 0.028	. 028	6	. 01840
. 029 . 030	0 3	1	3	1 2	1 }	0.029 to 0.030	. 030	3	. 02113
. 031 . 032	3	0	3 2	3	2 }	0.031 to 0.032	. 032	4	. 01460
. 033 . 034	3 2	0 4	0	3	1 }	0.033 to 0.034	. 034	2	. 01266
. 035 . 036	1 1	3	2 4	0	1 2	0.035 to 0.036	. 036	3	. 01829
. 037	0	1	4	3 4	1 2	0.037 to 0.038	. 038	3	. 02027
. 039 . 040	3 2	1 4	0	3	2 }	0.039 to 0.040	. 040	4	. 02299
Total								56	

DROPLET DISTRIBUTION

test, zero-gravity environment; vapor quality, 0.20; total weight flow rate, 0.0329 pound per second; local

Time for droplets of i th group mark to travel between two length marks in stream of vapor core, $(\Delta \tau)_{s,i}$,	f _i δ _{z,i} , droplets sec	Total droplets passing through reference point along con- denser tube, nt, i = nw, i + ns, i, droplets	Average number of droplets per unit volume of condenser tube, n _{v,i} , droplets	$\begin{aligned} &\text{drop-}\\ &\text{let flux,}\\ &\varphi_{i} = \frac{n_{t}}{A},\\ &\text{droplets} \end{aligned}$	n _{t, i} ^δ z, i	Total area of histogram, A _{t, i}
sec		sec	in. 3	(sec)(in. ²)		
0.00562	0.014	89	8	709	1.25	0.01465
. 00400	. 016	150	8	1194	2.40	. 02470
. 00595	. 054	324	24	2580	5. 83	. 05334
. 00727	. 080	448	32	3567	8.96	. 07376
. 00577	. 110	556	40	4427	12.23	. 09154
. 00580	. 144	730	48	5812	17.52	. 12018
. 00794	. 286	935	88	7444	24.31	. 15393
. 00585	. 168	676	48	5382	18.93	. 11129
. 00847	. 090	248	24	1975	7.44	. 04083
. 00454	. 128	578	32	4602	18.50	. 09516
. 00455	. 068	299	16	2381	10.17	. 04923
. 00493	. 108	386	24	3073	13.90	. 06355
. 00647	. 114	306	24	2436	11.63	. 05038
. 00763	. 160	349	32	2779	13.96	. 05746
	L	6074	448		167.03	1.00000

(f) Film 5 (60 in. from inlet); within interval of frames 3000 to 3600 counted from reference point; airplane test; vapor velocity, 7.17 feet per second.

Measured		Fr	ame	· · · · · ·	Measured	Average	Group	Absolute group-	Time for drop-
droplet diameter, $\frac{\delta}{z}$, in.	3000 Nu:	3200 mber o	3400 f dropl	3600 ets	frequency, f, number of droplets	group range, δ _z , in.	mark, δ _{z, i} , in.	mark frequency, f _i , number of droplets	lets of i th grou mark to travel between two length marks of
		1	· · · · · · · · · · · · · · · · · · ·						condenser wall $(\Delta \tau)_{W,i}$, sec
0.011	1 2	0	0	0	0 }	0.011 to 0.012	0.012	1	0.01458
.013	1	0	0 2	1	0	0.013 to 0.014	.014	1	. 00875
. 015	0 1	1 0	3 2	1	1 }	0.015 to 0.016	. 016	2	. 00972
. 017	2	3	4	0	2 }	0.017 to 0.018	. 018	3	. 00894
.018	0 2	3	1	2	1 J 2 }	0.019 to 0.020	. 020	4	. 01094
. 020 . 021	3 4	2 1	2 5	2	2 J 3 \	0.021 to 0.022	. 022	7	. 00948
. 022	6 3	3 2	2	5 2	4 J 3)				
. 024 . 025	4 5	6	1 7	2 4	3 } 5)	0.023 to 0.024	. 024	6	. 00719
.026	4 7	3 6	2	2	3 } 5)	0.025 to 0.026	. 026	8	. 00523
. 028	6	8	5	6	6	0.027 to 0.028	. 028	11	. 01878
. 030	8	3	5	7	6	0.028 to 0.030	. 030	10	. 03804
.031	3 5	2 4	3 7	5 3	3 5	0.031 to 0.032	. 032	8	. 01223
.033	2 1	5 0	6 5	4 6	4 3	0.033 to 0.034	. 034	7	. 02112
. 035	6 8	1 2	0 1	0 6	² }	0.035 to 0.036	. 036	6	. 02428
. 037	0 1	1 3	6 2	5 1	$\left.\begin{array}{c}3\\2\end{array}\right\}$	0.037 to 0.038	. 038	5	. 01311
. 039	2	0 4	1 0	0 2	1 }	0.039 to 0.040	. 040	3	. 02727
. 041	5 1	0 2	6 0	1	3 1	0.041 to 0.042	. 042	4	. 02800
Total			l	. !	- 1		l	86	

DROPLET DISTRIBUTION zero-gravity environment; vapor quality, 0.03; total weight flow rate, 0.0435 pound per second; local mean

Time for droplets of i th group mark	f _i o _{z, i} ,		Average number		$n_{t,i}^{\delta}_{z,i}$	Total area
	droplets	passing through	of droplets per	drop-		of histogram,
to travel between	sec	reference point	unit volume of	let flux,		$A_{t,i}$
two length marks		along con-	condenser tube,	$\varphi_{i} = \frac{n_{t}}{\Lambda}$		
in stream of		denser tube,	n _{v, i} ,	$\left \begin{array}{ccc} \psi_{\mathbf{i}} & \overline{\mathbf{A}}, \\ \overline{\mathbf{A}}, \end{array} \right $		
vapor core,		$n_{t,i} = n_{w,i} + n_{s,i}$	droplets	droplets		
$(\Delta \tau)_{s, i}$		droplets	in. 3	(sec)(in.2)		
sec		sec	ın.	(sec)(iii.)		
0.00273	0.012	158	8	1 258	1.90	0.01459
. 00219	. 014	217	8	1 728	3.04	. 02004
.00224	. 032	412	16	3 280	6. 59	. 03805
. 00275	. 054	562	24	4 475	10. 12	. 05191
. 00249	. 080	738	32	5 876	14. 76	. 06816
. 00349	. 154	1 119	56	8 909	24.62	. 10335
. 00248	. 144	1 311	48	10 438	31.46	. 12109
.00324	. 208	1 811	64	14 419	47.09	. 16727
. 00446	. 308	1 150	88	9 156	32.20	. 10622
. 00792	. 300	563	80	4 483	16.89	. 05200
. 00670	. 256	816	64	6 497	26.11	. 07537
. 00737	. 238	517	56	4 116	17.58	. 04775
. 00594	. 216	476	48	3 790	17.14	. 04396
.00641	. 190	501	40	3 989	19.04	. 04627
. 00763	. 120	195	24	1 553	7.80	. 01801
.00663	. 168	281	32	2 237	11.80	. 02595
		10 827	688		288. 14	0.99999

(g) Film 10 (59 in. from inlet); within interval of frames 3000 to 3600 counted from reference point; airplane vapor velocity, 7.17 feet per second.

Measured	0000		ame	1	Measured	Average	Group	Absolute group-	Time for drop-
droplet diameter, $\delta_{\mathbf{Z}}$, in.	3000 Nui	3200 mber o	3400 f drople	3600 ets	frequency, f, number of droplets	group range, $\delta_{\mathbf{z}}$, in.	mark, o z, i in.	mark frequency, f _i , number of droplets	lets of i th groumark to travel between two length marks condenser wall $(\Delta \tau)_{W, i}$,
0.017	0	1	0	T 1	0)				sec
.018	2	1	2	0	1	0.017 to 0.018	0.018	1	
. 019	1 2	1 1	3 0	0 2	1 }	0.019 to 0.020	. 020	2	
. 021	1	0	3	0	1 }	0.021 to 0.022	. 022	3	0.01034
. 022 . 023	3	3	2	3 2	2 J 3 \	0.023 to 0.024	. 024	5	. 01577
. 024 . 025	2 3	1 4	1 2	3 1	2 ∫ 3 }			,	
. 026	3	5	4	4	4 }	0.025 to 0.026	. 026	7	. 01441
.027	2 6	3 7	4 5	3 7	3 6	0.027 to 0.028	. 028	9	. 01500
. 029 . 030	7 9	5 10	4 6	3 8	4 }	0.029 to 0.030	. 030	12	. 01556
.031	3	5	4	7	5 }	0.031 to 0.032	. 032	11	. 02115
.032	4	3	5 2	7 6	6) 4 }	0.033 to 0.034	. 034	10	. 03382
.034	5 2	7	7 4	6 1	6 J 3]				
.036	3 6	8	6	2	5 }	0.035 to 0.036	. 036	8	.03060
.038	2	4	3	2	3	0.037 to 0.038	. 038	7	. 04153
. 039 . 040	4 3	1 2	2 4	2 2	3	0.039 to 0.040	. 040	5	. 03241
.041	5 3	1	4 2	3 2	$\left.\begin{array}{c} 3\\2\end{array}\right\}$	0.041 to 0.042	. 042	5	. 02917
.043	2	2	0	1	1 }	0.043 to 0.044	. 044	4	. 00636
.044	3 2	1	0	3	3 } 1 }	0.045 to 0.046	. 046	3	. 00959
. 046	3	1	2	2	2	2.22.00.010			. 00838
Total								92	

DROPLET DISTRIBUTION

test; 1-g environment; vapor quality, 0.03; total weight flow rate, 0.0435 pound per second; local mean

Time for droplets of i th group mark to travel between two length marks in stream of	f _i ^δ z, i', droplets sec	Total droplets passing through reference point along con- denser tube,	Average number of droplets per unit volume of condenser tube,	Average drop- let flux, $\varphi_i = \frac{n_t}{A},$	$n_{\mathrm{t,i}}^{\delta}_{\mathrm{z,i}}$	Total area of histogram, A _{t, i}
vapor core, $(\Delta \tau)_{s, i}$, sec		n _{t,i} =n _{w,i} +n _{s,i} , droplets sec	n _{v, i} , droplets in. 3	droplets (sec)(in.2)		
0.00124	0.018	242	8	1 927	4. 36	0.02294
. 00171	. 040	351	16	2 795	7.02	. 03327
. 00136	. 066	863	24	6 871	18.99	. 08181
. 00300	. 120	722	40	5 749	17. 33	. 06844
. 00339	. 182	960	56	7 644	24.96	. 09100
. 00300	. 252	1 320	72	10 510	36.96	. 12513
. 00327	. 360	1 640	96	13 058	49.20	. 15546
. 00478	. 352	1 054	88	8 392	33. 73	. 09991
. 00420	. 340	922	80	7 341	31.35	. 08740
. 00447	. 288	720	64	5 733	25. 92	. 06825
. 00420	. 266	618	56	4 921	23.48	. 05858
. 00600	. 200	358	40	2 850	14. 32	. 03394
	. 210	120	40	955	5.04	. 01138
	. 176	440	32	3 503	19.36	. 04171
	. 138	219	24	1 744	10.07	. 02076
		10 549	736	<u></u>	322.09	0.99998

(h) Film 6 (60 in. from inlet); within interval of frames 3200 to 3800 counted from reference point; airplane local mean vapor velocity, 5.94 feet per second

Measured		Fr	ame	·	Measured	Average	Group	Absolute group-	Time for drop-
droplet diameter, ⁸ z, in.	3200 Nur	3400 mber o	3600 f drople	3800 ets	frequency, f, number of droplets	group range, ⁸ z, in.	mark, ^δ z,i' in.	mark frequency, f _i , number of droplets	lets of i th group mark to travel between two length marks of condenser wall, $(\Delta \tau)_{\rm W,i}$, sec
0.017 .018	1 3	0	1 0	0	0 }	0.017 to 0.018	0.018	1	0.02400
. 019 . 020	2 1	0 1	4	3 2	2 1	0.019 to 0.020	. 020	3	. 02045
. 021 . 022	2 4	3 5	2 3	2 1	$\left. \begin{array}{c} 2 \\ 3 \end{array} \right\}$	0.021 to 0.022	. 022	5	. 02655
. 023 . 024	2 5	1 3	0 5	4	$\left.\begin{smallmatrix}2\\4\end{smallmatrix}\right\}$	0.023 to 0.024	. 024	6	. 01957
. 025 . 026	7 4	4 7	6 6	2 8	$\left.\begin{smallmatrix}4\\6\end{smallmatrix}\right\}$	0.025 to 0.026	. 026	10	. 02449
. 027 . 028	3 9	6 5	4 11	4 8	8	0.027 to 0.028	. 028	12	. 02509
. 029 . 030	3 8	3 5	2 3	4 3	5	0.029 to 0.030	. 030	8	. 02712
. 031	3 5	2 3	4 6	5 4	3 4	0.031 to 0.032	. 032	7	. 01522
. 033	5	1	4 1	5 0	$\left.\begin{array}{c} 3\\2\end{array}\right\}$	0.033 to 0.034	. 034	5	. 01288
. 035	3 4	2 0	2 5	1	$\left.\begin{smallmatrix}2\\2\end{smallmatrix}\right\}$	0.035 to 0.036	. 036	4	. 01905
. 037	3 2	4 0	2 1	2 1	$\left.\begin{array}{c} 3\\1\end{array}\right\}$	0.037 to 0.038	. 038	4	. 01678
. 039	0 3	3	0 2	0 2	1 }	0.039 to 0.040	. 040	3	. 01957
. 041	1 0	3 1	0 0	1 2	1 }	0.041 to 0.042	. 042	2	. 01558
Total								70	

DROPLET DISTRIBUTION

test; zero-gravity environment; vapor quality, 0.03; total weight flow rate, 0.0315 pound per second;

				+		
Time for droplets	f _i δ _{z,i} ,	Total droplets	Average number	Average	n t, i $^{\delta}$ z, i	Total area
of i th group mark	droplets	passing through	of droplets per	drop-		of histogram,
to travel between	sec	reference point	unit volume of	let flux,		A _{t,i}
two length marks		along con-	condenser tube,	$\varphi_{\mathbf{i}} = \frac{\mathbf{n_t}}{\mathbf{A}},$		
in stream of		denser tube,	n _{v,i} ,	$ \varphi_{\mathbf{i}} = \overline{\mathbf{A}}' $		
vapor core,		n _{t,i} =n _{w,i} +n _{s,i} ,	droplets	droplets		
$(\Delta au)_{\mathbf{S},\mathbf{i}},$		droplets	in. 3	(sec)(in. ²)		
sec		sec	III.	(Sec)(III.)		
0.00702	0.018	82	8	653	1.48	0.01371
	0.025					·
. 1043	. 060	203	24	1616	4.06	. 03395
. 00541	. 110	483	40	3846	10.63	. 08078
. 00882	. 144	456	48	3631	10.94	. 07627
. 00983	. 260	652	80	5191	16.95	. 10905
. 00966	. 336	784	96	6242	21.95	. 13113
. 00426	. 240	929	64	7397	27.87	. 15538
00540				0014	07.00	10000
. 00542	. 224	793	56	6314	25. 38	. 13263
. 00694	. 170	521	35	4148	17. 71	. 08714
				ļ		
00700		994	20	0050	10.00	. 05586
. 00769	. 144	334	32	2659	12.02	.05560
. 00667	. 152	383	32	3049	14. 55	. 06406
. 00816	. 120	239	24	1903	9.56	. 03997
. 01860	. 084	120	16	955	5.04	. 02007
		l				
		5979	555		178. 14	1,00000
		3313		<u> </u>		1.0000

(i) Film 11 (59 in. from inlet); within interval of frames 3200 to 3800 counted from reference point; airplane vapor velocity, 11.60 feet per second

Measured		Fra	ame		Measured	Average	Group	Absolute group-	Time for drop-
droplet diameter, $\delta_{_{f Z}},$ in.	3200 Nu	3400 mber o	3600 f dropl	3800 ets	frequency, f, number of droplets	group range, 6 ₂ , in.	mark, ⁸ z, i' in.	mark frequency, f _i , number of droplets	lets of i th group mark to travel between two length marks of condenser wall, (Δτ) _{w,i} , sec
0.021	1	0	1	0	0 }	0.021 to 0.022	0.022	1	
. 022	2	1	0	1	•				
. 023 . 024	0	0	1	0 2	$\begin{bmatrix} & 1 \\ & 1 \end{bmatrix}$	0.023 to 0.024	. 024	2	
. 024	3	1	0	0	1)				
. 026	4	2	5	1	3	0.025 to 0.026	. 026	4	
. 027	3	2	1	1	2)				
. 028	1	2	1	1	1	0.027 to 0.028	. 028	7	
. 029	3	1	0	0	1]	0 000 4 0 000	000		
. 030	4	2	3	4	3 }	0.029 to 0.030	. 030	9	
. 031	0	3	2	3	2 }	0.031 to 0.032	. 032	11	0. 02933
. 032	3	4	6	4	4)	0.502.00.002	7002		
. 033	2	4	3	3	3 }	0.033 to 0.034	. 034	8	. 03122
. 034	1	0	4	3	2 }				
. 035 . 036	1 2	3	1 2	1 1	$\left\{\begin{array}{c} 1 \\ 2 \end{array}\right\}$	0.035 to 0.036	. 036	6	. 04286
. 037	2	0	1	0	1)				
. 038	2	3	2	4	3	0.037 to 0.038	. 038	4	. 02254
. 039	3	2	1	2	2]				
. 040	0	0	2	1	1	0.039 to 0.040	. 040	3	. 00830
. 041	3	0	1	1	1]	0.041 4.0.040	045		00004
. 042	0	0	2	2	1 }	0.041 to 0.042	. 042	2	. 00884
. 043	0	1	1	1	1 }	0.043 to 0.044	. 044	1	. 00734
. 044	3	0	0	2	1]			•	
. 045	1	1	0	2	1 }	0.045 to 0.046	. 046	2	. 01143
. 046	0	1	1	1	1)				
Total								60	

DROPLET DISTRIBUTION

test; 1-g environment; vapor quality, 0.03; total weight flow rate, 0.0324 pound per second; local mean

Time for droplets of $\frac{1}{10}$ group mark to travel between two length marks in stream of vapor core, $(\Delta \tau)_{s,1'}$ sec $\frac{1}{10}$ models $\frac{1}{10}$ sec $\frac{1}{10}$ models $\frac{1}{10}$ sec $\frac{1}{10}$ models $\frac{1}{10}$ sec $\frac{1}{10}$ models $\frac{1}$							
to travel between two length marks in stream of vapor core, (\(\lambda \cdot \) accepted by a length marks in stream of vapor core, (\(\lambda \cdot \) accepted by a length marks sec. 0.00323	Time for droplets	f _i δ _{z,i} ,	Total droplets			$^{n}_{t,i}^{\delta}_{z,i}$	Total area
Teference point Instrument		droplets		_	-		
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$							A _{t, i}
vapor core, $(\Delta \tau)_{8,1}$? sec. $n_{t,1} = n_{w,1} + n_{8,1}$? $\frac{d \operatorname{roplets}}{\operatorname{sec}}$ $\frac{d \operatorname{roplets}}{\operatorname{in.}}$ $\frac{d \operatorname{roplets}}{\operatorname{(sec)(in.}^2)}$ 0.00323 0.022 62 8 494 1.36 0.01298 .00339 .048 118 16 940 2.83 .02471 .00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .084 181 16 1441 7.60 .03790 .044	_			· 1	$\omega = \frac{n_t}{t}$		
(AT) _{8,1} ? sec droplets sec in. 3 (sec)(in. 2) 0.00323 0.022 62 8 494 1.36 0.01298 .00339 .048 118 16 940 2.83 .02471 .00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282 <				n _{v,i} ,	$\begin{bmatrix} \mathbf{v_1} & \mathbf{A} \end{bmatrix}$		
sec sec <td>· -</td> <td></td> <td>$n_{t,i} = n_{w,i} + n_{s,i}$</td> <td></td> <td>droplets</td> <td></td> <td></td>	· -		$n_{t,i} = n_{w,i} + n_{s,i}$		droplets		
sec sec 0.00323 0.022 62 8 494 1.36 0.01298 .00339 .048 118 16 940 2.83 .02471 .00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	$\left(\Delta\tau\right)_{s,i}$		droplets	in. 3	$(sec)(in.^2)$		
.00339 .048 118 16 940 2.83 .02471 .00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	sec		sec		(===,(===,		
.00339 .048 118 16 940 2.83 .02471 .00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
.00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	0.00323	0.022	62	8	494	1.36	0.01298
.00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
.00295 .104 271 32 2158 7.05 .05674 .00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282					0.40		00.451
.00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	. 00339	. 048	118	16	940	2.83	. 02471
.00289 .196 485 56 3862 13.58 .10155 .00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
.00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	. 00295	. 104	271	32	2158	7.05	. 05674
.00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282						1	
.00292 .270 616 72 4905 18.48 .12898 .00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282			40-		0000	10.50	10155
.00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	. 00289	. 196	485	56	3862	13.58	. 10155
.00391 .352 863 88 6871 27.62 .18070 .00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
.00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	. 00292	. 270	616	72	4905	18.48	. 12898
.00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
.00317 .272 710 64 5653 24.14 .14866 .00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	00201	250	062		6971	27.62	18070
.00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	.00351	. 352	803	00	0011	21.02	. 100.0
.00300 .216 512 48 4077 18.43 .10720 .00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282		j					
.00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	. 00317	. 272	710	64	5653	24.14	. 14866
.00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282						ŀ	
.00288 .152 420 32 3344 15.96 .08794 .120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282	.00300	216	512	48	4077	18.43	. 10720
.120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282			, , , , , , , , , , , , , , , , , , ,				
.120 289 24 2301 11.56 .06051 .084 181 16 1441 7.60 .03790 .044 109 8 868 4.80 .02282							
084 181 16 1441 7.60 .03790 044 109 8 868 4.80 .02282	. 00288	. 152	420	32	3344	15.96	. 08794
084 181 16 1441 7.60 .03790 044 109 8 868 4.80 .02282						1	
084 181 16 1441 7.60 .03790 044 109 8 868 4.80 .02282		. 120	289	24	2301	11.56	. 06051
.044 109 8 868 4.80 .02282							
.044 109 8 868 4.80 .02282						F 22	00700
		.084	181	16	1441	7.60	. 03790
092 140 16 1115 6.44 .02931		. 044	109	8	868	4.80	. 02282
092 140 16 1115 6.44 .02931							
.022 140 10 1110 6.44 .0231		000	140	10	1115	6.44	02031
		. 092	140	15	1113	1 2.22	. 02501
1 1 1 1 1 1 1 1		<u> </u>		 	 	150.05	1 00000
4776 480 159.85 1.00000			4776	480		159.85	1.00000

(j) Film 14 (47 in. from inlet); within interval of frames 3400 to 3700 counted from reference point; airplane vapor velocity, 43.00 feet per second

Measured		Fr	ame		Measured	Average	Group	Absolute group-	Time for drop-
droplet diameter, δ_z , in.	3400 Nur	00 2200 3600 2400 3 0 0				mark, δ _{z,i} , in.	mark frequency, f ₁ , number of droplets	lets of i th group mark to travel between two length marks of	
									condenser wall, $(\Delta \tau)_{W, 1}$, sec
0.019	1	0 2	1 0	0	0 }	0.019 to 0.020	0.020	1	0.01014
. 021 . 022	0	2	1 0	1 2	1 }	0.021 to 0.022	. 022	2	. 01029
. 023 . 024	0	1 2	1 3	1 1	1 }	0.023 to 0.024	. 024	3	. 01963
. 025 . 026	2	2	4	1 2	$\left.\begin{array}{c}2\\3\end{array}\right\}$	0. 025 to 0. 026	. 026	5	. 03500
. 027 . 028	0	4	2 4	3 2	2 4	0. 027 to 0. 028	. 028	6	. 01359
. 029 . 030	1 4	5 2	4	2	$\begin{pmatrix} 3 \\ 4 \end{pmatrix}$	0. 029 to 0. 030	. 030	7	. 01591
. 031 . 032	7 3	1 7	3	4	4 }	0. 031 to 0. 032	. 032	8	. 01538
. 033 . 034	2	2	4	3	3 4	0. 033 to 0. 034	. 034	7	. 01828
. 035 . 036	3 5	2	3 1	3	3 }	0.035 to 0.036	. 036	6	. 01321
. 037 . 038	1 2	2	2 3	4	2 3	0.037 to 0.038	. 038	5	. 03571
. 039 . 040	0	1 2	2 5	1 2	1 3	0.039 to 0.040	. 040	4	. 03373
. 041 . 047	0 2	1 3	1 2	3	1 }	0.041 to 0.044	. 044	3	. 04200
Total	<u></u>		<u> </u>	l		<u>,</u>	L	57	

DROPLET DISTRIBUTION

test: 1-g environment; vapor quality, 0.20; total weight flow rate, 0.0329 pound per second; local mean

Time for droplets of i th group mark to travel between two length marks in stream of vapor core, $(\Delta \tau)_{8,i}$, sec	f _i δ _{z,i} , droplets sec	Total droplets passing through reference point along con- denser tube, nt, i = nw, i + ns, i, droplets sec	Average number of droplets per unit volume of condenser tube, nv,i, droplets in. 3	drop- let flux,	n _{t,i} ō _{z,i}	Total area of histogram, ^A t, i
0.00390	0.020	146	8	1162	2.92	0.03023
. 00522	. 044	251	16	1998	5. 52	. 05197
. 00826	. 072	216	18	1720	5.18	. 04472
. 00698	.130	315	40	2508	8. 19	. 06522
. 00756	. 168	547	48	4355	15. 32	. 11325
. 00442	.210	783	56	6234	23.49	. 16211
. 00842	. 256	649	64	5167	20.77	. 13437
. 00732	. 238	555	56	4419	18. 87	. 11491
. 00522	. 216	663	48	5279	23.87	. 13727
. 00644	. 190	331	40	2635	12.58	. 06853
. 00851	. 160	224	32	1783	8.96	. 04638
. 00900	. 132	150	24	1194	6.60	. 03106
		4830	450		152. 27	1.00002

per second passing through a reference point along the condenser tube was obtained from the relation

$$n_{t,i} = n_{w,i} + n_{s,i}$$

$$= f_i \left(\frac{\text{Percentage of drops on wall}}{\Delta \tau_{w,i}} + \frac{\text{Percentage of drops in stream of vapor core}}{\Delta \tau_{s,i}} \right)$$
(4)

where, $\Delta \tau_{w,i}$ and $\Delta \tau_{s,i}$ are the time intervals elapsed for a droplet size of the i^{th} group mark to travel between the two length marks on a condenser wall and in a stream of vapor core, respectively.

The number of droplets obtained from equation (4) is called absolute frequency per second of the i^{th} interval of group mark of the droplet size. In order to obtain the number of droplets per unit volume of a condenser tube, the absolute group-mark frequency f_i of a given droplet size was divided by the condenser volume confined between the two length marks. Thus,

$$n_{v,i} = \frac{f_i}{\frac{\pi}{4} d^2 l}$$
 (5)

Dividing the absolute droplet frequency per second (given in eq. (4)) by the cross-sectional area of a condenser tube gives the droplet flux (number of droplets per unit cross-sectional area per unit time). Thus,

$$\varphi_{i} = \frac{f_{i}}{\frac{\pi}{4}} \frac{\left(\frac{\text{Percentage of droplets on wall}}{\Delta \tau_{w, i}} + \frac{\text{Percentage of droplets in stream of vapor core}}{\Delta \tau_{w, s}}\right)}{\Delta \tau_{w, s}}$$
(6)

The measured values were reduced by using equations (1) to (6) for each interval of droplet size analyzed. The data, thus, obtained are presented in table III. In order to translate the data given in these tables into graphic form, a system of rectangle coordinates was used, as outlined in reference 5. The abscissa represents the variable droplet diameter δ_i , and the ordinate the absolute frequencies per group interval. The rectangle heights for $n_{t,i}$ and $n_{v,i}$ are defined, respectively, by

$$h_{t,i} = \frac{n_{t,i}}{\Delta \delta \sum_{i=1}^{m} n_{t,i}} = \frac{n_{t,i}}{\Delta \delta N_t}$$
(7)

$$h_{v,i} = \frac{n_{v,i}}{\Delta \delta \sum_{i=1}^{m} n_{v,i}} = \frac{n_{v,i}}{\Delta \delta N_{v}}$$
(8)

which denote the absolute frequencies centered at δ_i , for instance, and were calculated for each interval of droplet size. When the ordinates of the rectangles $h_{t,i}$ and $h_{v,i}$ were multiplied by the base width $\Delta\delta$, the areas of the respective rectangles were obtained. Thus,

$$A_{t,i} = h_{t,i} \Delta \delta = \frac{n_{t,i}}{N_t}$$
 (9)

$$A_{v,i} = h_{v,i} \Delta \delta = \frac{n_{v,i}}{N_v}$$
 (10)

The area calculated from these equations represents the relative frequencies centered at δ_i . The total calculated area of any histogram approached 1 closely, because the sum of the relative frequencies must be equal to 1. For example, at the vapor quality of 0.03, a sum of the areas of all calculated rectangles of the histogram, as tabulated

in table III(d), is $\sum_{i=1}^{m} A_{t,i} = 0.9995$, whereas the theoretical value, as defined by equa-

tion (B4) in appendix B, is 1. The data calculated by using equations (7) to (9) were tabulated in table III, and the results are plotted in figures 10 to 13, respectively. In these figures, the height of the rectangles, which represents the absolute frequencies, and therefore, the areas which represent the corresponding relative frequencies of the histogram are only proportional to the actual calculated values rather than equal to them.

With a convenient choice of a rectangle width $\Delta\delta$ of 1 unit on the δ -axis, the numerical values obtained for the rectangle heights from equations (7) and (8) are equal to the corresponding rectangle areas, calculated from equations (9) and (10), respectively.

TABLE IV. - MEASURED AND CALCULATED PERTINENT PARAMETERS OF DROPLET SIZE DISTRIBUTION

IN CONDENSER TUBE IN GROUND TESTS

	N(8 _{mf})8 _{mf}			8.109	15.400	15.930	35, 310		1.057	1. 265	1. 796	3. 200
	(b ^{b+1})e -b			111 823.6844	207 961. 0364	3 644. 3118	80 192. 851		12 685 067. 3664	30, 1085	902. 1847	794 094. 9154
Calculated	r(b + 1)			13 762.9147	13 496.9520	229. 2018	2 273. 0400		12 012 374, 4000	23. 8162	502, 8862	247 844. 8550
Ü	Most frequent	droplet diameter, $\delta_{mf} = b/c$	time	0.01254	. 01681	. 01406	. 01854	rolume	0. 01079	. 01499	. 01373	. 01863
	$N(\delta_{mf})^{\delta}mf$ Characteristic parameters in eq. (B2)	ပ	Number of droplets per unit time	0.2764	.4690	25.0660	5.1470	Number of droplets per unit volume	0.6000×10 ⁻⁴ 0.01079	14.4500	1.4130	.4650×10 ⁻²
	eristic par eq. (B2)	q	er of dr	7. 499	7.485	5.372	6.600	er of dro	10.504	3.996	5.807	8. 830
	Charact	ત	Numk	598.10	445.08	381.98	355.97	Numbe	972. 81	266.49	422.93	473.74
	N(8mf)8mf			8. 125	15.408	15.900	35.280		1.056	1. 264	1. 794	3, 204
Measured	Most frequent	aropiet diameter, δmf	-	0.013	910 .	. 015	. 018		0.011	. 016	. 013	. 018
	quality, Maximum x value of	gamma distribu- tion function, N($\delta_{ m mf}$)		625	963	1060	1960		96	79	138	178
Vapor	quality, x			0.82	. 36	. 20	. 03		0. 82	. 36	. 20	ස
Film				-	8	ო	4		1	8	ო	4

Inspection of histograms of the droplet size distribution showed a marked similarity in shape with gamma distribution given in reference 6. This suggested the choice of the following three-parameter frequency distribution function,

$$N(\delta) = C(a\delta)^{b} e^{-a\delta}$$
 (11)

which is derived from the gamma probability density function in appendix B. The three parameters that comprise a frequency function $N(\delta)$ were determined from the measured data in conjunction with equations (B7) to (B9). A digital computer and a FORTRAN program (Least-Square Estimation of Nonlinear Parameters described in ref. 7, which is available as IBM Share Program No. 1428) were used for this purpose. The numerical values of three characteristic parameters thus determined are tabulated in table IV for ground tests (1-g environment) and in table V for airplane tests (1-g and zero-gravity environments). By substituting these values of the representative parameters into equation (11), a set of two equations, for each vapor quality condition analyzed, may be obtained. Each equation expresses the absolute droplet frequency $N(\delta)$ per unit time and per unit volume, respectively, having a certain diameter δ .

Correlation of Parameters a, b, and c and δ_{mf} With System Parameters

In order to obtain a generalized expression for predicting droplet size distribution in a condenser tube, the values of three parameters a, b, and c, and the most frequent droplet size δ_{mf} necessary in describing the gamma distribution law were correlated as a function of system parameters such as mean vapor velocity $V_{m,\,v}$ and vapor quality x. In a range of system parameters investigated $4 \le V_{m,\,v} \le 129$ feet per second and $0.03 \le x \le 0.82$, these parameters were correlated for two conditions of droplet distribution: for expressing the absolute droplet frequency per unit time and per unit volume of a condenser tube.

Ground tests. - In the ground test analysis, the inlet vapor velocity was 141 feet per second, the inlet vapor quality 90 percent or greater, and the total mercury weight flow rate was 0.0385 pound per second. The vapor velocity distribution along the condenser tube is discussed in appendix C.

For a condition where the time function was included in the analysis, the following empirical equations were obtained:

$$a_{t} = (V_{m,v})x^{0.069} + 10^{2.559}$$
 (12)

$$b_{t} = a_{t,i} \left(\delta_{mf} \right)_{t} \tag{13}$$

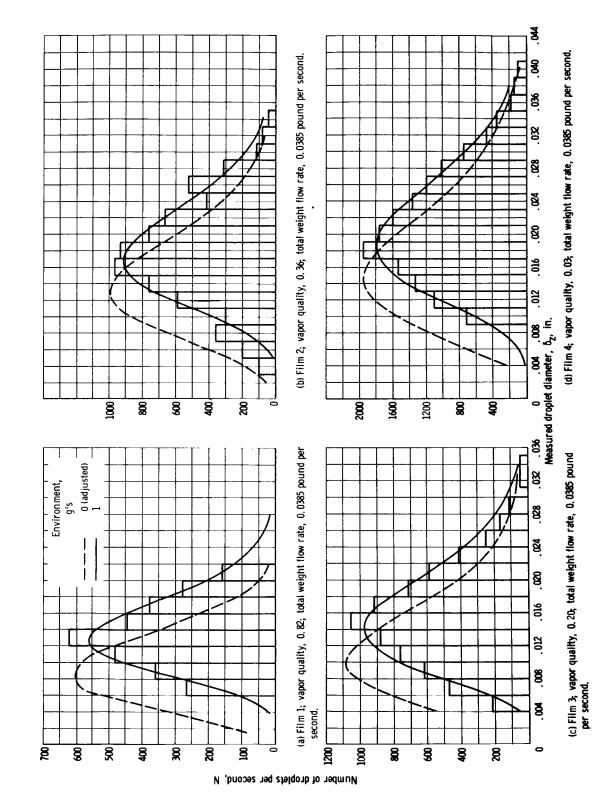


Figure 10. - Distribution of droplet size per unit time in ground tests.

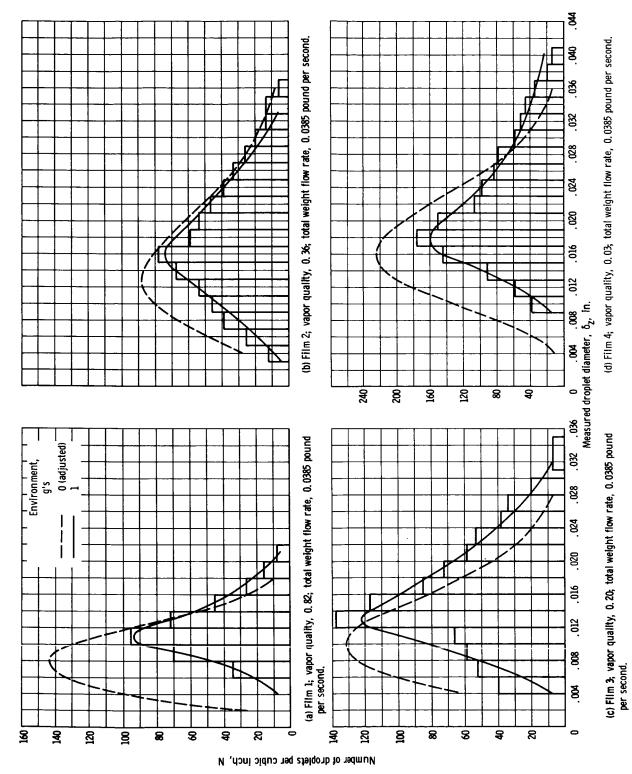


Figure 11. - Distribution of droplet size per unit volume in ground tests.

$$c_{t} = \frac{b_{t}}{\left(\delta_{mf}\right)_{t}} \frac{1}{\Gamma\left(b_{t} + 1\right)}$$
 (14)

$$\left(\delta_{\mathrm{mf}}\right)_{\mathrm{t}} = 5684.620 - 58.912(1 - x)V_{\mathrm{m,v}}$$
 (15)

For a condition where the volume function was included in the analysis, the correlated parameters were expressed by the following empirical equations:

$$a_v = 0.170(V_{m,v} - x)^2 - 13.328 V_{m,v} x + 479.400$$
 (16)

$$b_{v} = a_{v} \left(\delta_{mf} \right)_{v} \tag{17}$$

$$c_{v} = \frac{b_{v}}{\left(\delta_{fm}\right)_{v}} \frac{1}{\Gamma\left(b_{v} + 1\right)}$$
 (18)

$$\left(\delta_{\rm mf}\right)_{\rm v} = 5684.620 - 58.912(1 - x)V_{\rm m, v}$$
 (19)

By substituting the values of the correlated parameters into equation (11), the two general empirical equations were obtained for predicting droplet size distribution in a condenser tube, which gave the absolute frequencies of the droplet size distribution (the number of droplets of a given size per second, and the number of droplets of a given size per cubic inch of a condenser tube). These two equations thus obtained are plotted in figures 10 and 11. An accuracy of correlation of the measured data with the empirical equation (11) used, gave a deviation of 1.36 rms percent for the plots shown in figure 10, and 1.72 rms percent for the plots shown in figure 11.

Airplane tests. - The airplane tests were conducted in 1-g and zero-gravity environments. In these tests, the vapor inlet velocity ranged from 165 to 235 feet per second, the vapor inlet quality was 90 percent or greater, and the total mercury weight flow rate ranged from 0.0315 to 0.0435 pound per second. In the airplane tests, the films of a condensing mercury vapor flow were taken simultaneously at two different locations (two high-speed cameras were available) for vapor quality conditions of 0.20 and 0.03.

The field of view of the camera in the airplane installation was limited to about 4 inches of the condensation tube. Therefore, the droplet magnification from film to the analyzer screen, depending on magnification lens used, ranged from 1.4 to 2.2. Because of the small magnification factor, small droplets were not visible in the condenser tube;

TABLE V. - MEASURED AND CALCULATED PERTINENT PARAMETERS OF DROPLET SIZE DISTRIBUTION

IN CONDENSER TUBE IN AIRPLANE TEST

(a) Total number of droplets per unit time (sec) passing through reference point in condenser tube

									_	
	_b b+1 e-b			4560	4100	4030		4718	4740	3170
	r(b + 1) b ^{b+1} e ^{-b}			636. 780	577. 080	568.690		655. 770	658.090	642. 030
Calculated	Most frequent	droplet diameter, $\delta_{mf} = (b/a) + d$		0.026	. 030	. 025		0: 030	. 032	. 029
Ca	eters	ъ		0.006	. 010	900 .		0.010	. 012	. 010
	param	ပ		2.234	1.370	1.342		1.979	1.094	1.409
	teristic	Q	nent	6.935	6.882	6.874		6.950	6.952	6. 758
	Characteristic parameters	ત	environ	354, 899 6, 935 2, 234 0, 006	349.998 6.882 1.370	355.990 6.874 1.342	1-g environment	356.000 6.950 1.979 0.010	355.990 6.952 1.094	350, 030 6, 758 1, 409
	N(8mf)8mf		Zero-gravity environment	47.086	27.870	24.310	1-g envi	49.200	27.620	23.490
Measured	value of Most frequent N(8 _{mf})8 _{mf}	droplet diameter, $^{\delta}$ mf	2	0.026	. 030	. 026		0.030	. 032	. 030
W	Martimum	garina distribution function, $N(\delta_{mf})$		1811	929	935		1640	863	783
Vapor	quality, x			0.03	. 03	. 20		0.03	. 03	. 20
Film				2	9	2		8	6	10

(b) Ratios of calculated parameters (0 to 1 g) necessary to describe droplet size distribution (for $n_{t,\, i}$)

Film	Vapor	:		Gravity factors	ctors	
	quality, x	$A_t = a_0/a_1$	$B_t = b_0/b_1$	$c_t = c_0/c_1$	$D_t = d_0/d_1$	quality, $A_t = a_0/a_1$ $B_t = b_0/b_1$ $C_t = C_0/C_1$ $D_t = d_0/d_1$ $\Delta_t = \left[\left(\delta_{mf} \right) / \left(\delta_{mf} \right) \right]$
5,8	0.03	0.9960	0.9978	1.1287	0.8666	0. 601
6,9	. O3	.9831	6686	1.2520	. 9375	. 833
7,10	. 20	1.0170	1.0170	. 9528	9998 .	009
Root-mean	nean	1.000	1.000	1.111	0.880	0.705
sdna	square value					

TABLE V. - Concluded. MEASURED AND CALCULATED PERTINENT PARAMETERS OF DROPLET SIZE

DISTRIBUTION IN CONDENSER TUBE IN AIRPLANE TEST

(c) Total number of droplets per unit volume (in. 3) passing through reference point in condenser tube

\Box	q									
	_b b+1 е-			5810	628	1650		2290	1025	3873
	Γ(b + 1) b ^{b+1} e ^{-b}			787.630	111.030	257.860		342. 730	161.930	541.030
Calculated	ent	droplet diameter, $\delta_{mf} = (b/a) + d$		0.028	. 029	. 026		0.030	. 031	. 032
Ca	eters	ਚ		900 0	. 010	800 .		0.012	. 012	. 012
	param	ວ		0.098	. 745	. 268		0. 253	.470	. 104
	teristic	Q	nent	7. 048	5.955	6.438		6.600	6.174	6.847
	Characteristic parameters	ત્ય	environn	353.979 7.048 0.098 0.008	348.047 5.955	350. 009 6. 438	1-g environment	355.998 6.600 0.253 0.012	356.041 6.174	350.000 6.847
	N(5mf) ⁵ mf		Zero-gravity environment	1.664	2.688	2.288	1-g env	2.880	2.816	2.048
Measured	value of Most frequent N(6 _{mf})6 _{mf}	dropiet diameter, ⁵ mf		0. 026	. 028	. 026		0.030	. 032	. 032
	quality, Maximum value of	gamma distribution function, $N(\delta_{mf})$		64	96	88		96	88	64
Vapor	quality,			0.03	. 03	. 20		0.03	.03	. 20
Film				2	မ	7		8	o	91

(d) Ratios of calculated parameters (0 to 1 g) necessary to describe droplet size distribution (for $n_{t,\;j})$

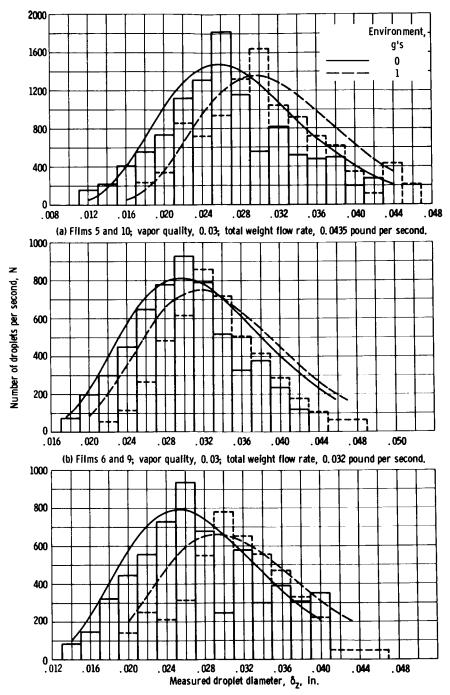
	quality, $A_v = a_0/a_1 B_v = b_0/b_1 C_v = C_0/C_1 D_v = d_0/d_1 A_v = (\delta_{mf})/(\delta_{mf})$	0.867 .875 .812	0.831
ctors	$D_{v}=d_{0}/d_{1}$	0.667 .833 .667	0. 723
Gravity factors	$c_v = c_0/c_1$	0.387 1.588 2.572	1.516
	$\mathbf{B_v} = \mathbf{b_0}/\mathbf{b_1}$	1.068 .964 .940	0.997
	$A_{\rm v}=a_0/a_1$	0.994 .978 1.000	0.998
Vapor	quality, x	0.03	ot-mean- square value
Film		5,8 6,9 7,10	Root-mean- square va

therefore, the droplet size distribution obtained from the airplane tests indicates mainly the effect of gravity level rather than an actual droplet size distribution.

The numerical values of the three characteristic parameters a, b, and c were determined for each test run and gravity level investigated from the test data. The results obtained are tabulated in table V for 1-g and zero-gravity environments for expressing the absolute droplet frequency per unit time and the number of droplets per unit volume of a condenser tube. By substituting these values into equation (11), two empirical equations were obtained for each test run, which gave the absolute frequencies of the droplet size distribution, the number of droplets of a given size per second, and the number of droplets of a given size per second, and the number of droplets of a given size per cubic inch of condenser tube. These two equations are plotted in figures 12 and 13 for 1-g and zero-gravity environments, respectively. The accuracy of the measured data relative to the empirical equation (eq. (11)) obtained in the airplane tests ranged from 1.83 to 2.40 rms percent.

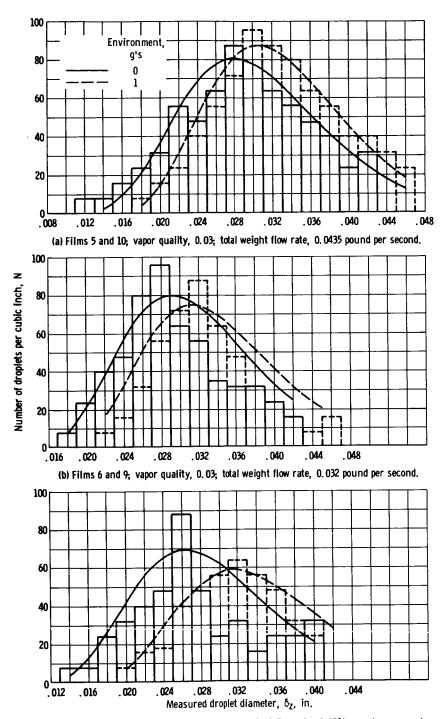
Effect of gravity. - The effect of gravity level on the droplet size distribution in a condenser tube can be seen from figures 12 and 13. In zero gravity, mercury droplets were more uniformly dispersed through the vapor phase. In addition to this, observation indicated that in the vapor stream and on the condenser wall, the droplets were moving toward the liquid interface, with more uniform axial velocity (less turbulent) in zero gravity than in a 1-g environment. On the average, the size of droplets were somewhat larger and lesser in number in a 1-g than in a zero-gravity environment. For example, for the same total weight flow rate and vapor quality condition, the most frequent droplet diameter in 1 g was 0.030 inch, and in zero gravity 0.026 inch (see fig. 14). In zero gravity, droplet distribution was more symmetric about its most frequent droplet size than in a 1-g environment. The larger droplets, in a 1-g environment, on the tube bottom resulted from the effect of agglomeration, as the droplets migrated toward the bottom of the tube. Figure 15 indicates the degree to which gravity affects the flow regime. At 1 g, the droplets on the wall have flowed to the bottom of the tube, agglomerating into larger droplets. In zero-gravity environment, the droplets on the wall have no tendency to migrate and collect in any one part of the tube wall, rather they flow parallel to the tube axis.

As can be seen from figures 12 and 13, in the range of mercury weight flow rate and vapor quality conditions studied, the pattern of droplet size distribution varied in the same fashion, approximately, from zero gravity to a 1-g environment. Furthermore, examination of table V indicates that the ratios (\mathscr{A} , \mathscr{A} , \mathscr{C} , and Δ) of the parameter necessary to describe the droplet size distribution law for zero gravity to that of a 1-g environment varied little from one test condition to the other. Therefore, in order to transpose the droplet size distribution from 1-g to zero-gravity environment, it is necessary to correct the parameters a, b, c, and δ_{mf} of 1-g environment by the proper gravity factors \mathscr{A} , \mathscr{A} , and Δ , respectively.



(c) Films 7 and 10; vapor quality, 0.02; total weight flow rate, 0.0329 pound per second.

Figure 12. - Distribution of droplet size per unit time in airplane tests.



(c) Films 7 and 10; vapor quality, 0.20; total weight flow rate, 0.0329 pound per second. Figure 13. - Distribution of droplet size per unit volume in airplane tests.

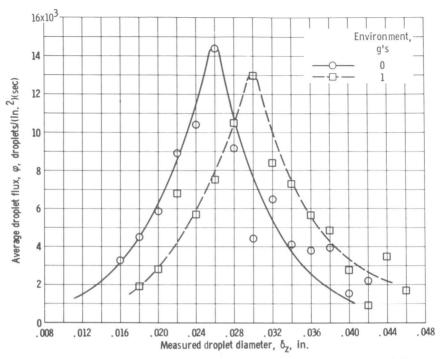
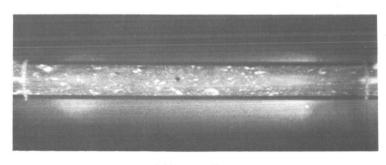
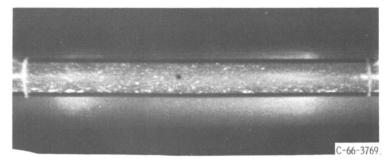


Figure 14. - Average distribution of droplet flux obtained from films 5 and 10. Airplane test; total weight flow rate, 0.0435 pound per second; vapor quality, 0.03.



(a) Zero gravity.



(b) 1 g.

Flow direction

Figure 15. - Condensing mercury vapor flow in vapor region. Vapor quality condition 0.20, total weight flow rate, 0.0435 pound per second; tube diameter, 0.400 inch.

Modification of Ground Test Data to Obtain Zero-Gravity Droplet Distribution

In order to define a droplet size distribution function in zero-gravity environment from the ground test data, the parameters a, b, c, and δ_{mf} expressed by equations (12) to (18), necessary to describe the droplet size distribution law on the ground, were corrected by the gravity factors \mathscr{A} , \mathscr{A} , and Δ , respectively. The values of these factors are listed in table V (pp. 41 and 42).

By substituting the values of the corrected parameters into equation (11), two empirical equations were obtained for predicting droplet size distribution in a condenser tube in zero-gravity environment when the ground test data are known. One of the equations expressed the number of droplets of a given size per second, and the other, the number of droplets of a given size per cubic inch of a condenser tube. These two equations, thus obtained, are plotted in figures 10 and 11.

<u>Droplet flux.</u> - The expression for droplet flux, given by equation (6), defines the number of droplets of a given size crossing through the cross-sectional area of a condenser tube per unit area per unit time.

Ground tests. - Equation (6) was used to calculate the droplet flux for each interval of the droplet size analyzed. The data obtained are tabulated in table III (pp. 13 to 33), and the results are presented in figures 16 and 17. In figure 16, the droplet flux obtained by cross-plotting figure 14 for the three most frequent droplet diameters (0.010, 0.015, and 0.020 in.) in a condenser tube is plotted as a function of a droplet diameter for various

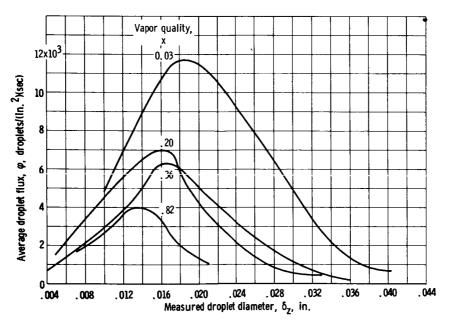


Figure 16. - Average distribution of droplet flux obtained from films 1 to 4. Ground test; total weight flow rate, 0.0385 pound per second.

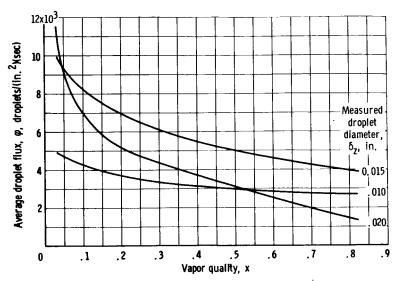


Figure 17. - Average distribution of droplet flux as function of vapor quality. Ground test; total weight flow rate, 0.0385 pound per second.

vapor qualities. As shown in this plot, the droplet flux increases very fast with a decrease in vapor quality. For example, an increase in the droplet flux of the 0.015-inch-diameter droplet, with a decrease in vapor quality is shown in the following table:

Vapor quality,	Droplet flux,
x	φ ,
	Number of droplets
	(in. ²)(sec)
0.80	2700
. 50	5000
. 03	9900

Effect of gravity on droplet flux. - In a range of mercury weight flow rate and the vapor quality conditions investigated, equation (6) was used to calculate the average droplet flux for the airplane tests. The data obtained for 1-g and zero-gravity environments are tabulated in tables III(e) to (i)(pp. 22 to 31).

For the same mercury weight flow rate and vapor quality conditions investigated, the effect of gravity environment on the droplet flux in a condenser tube is shown in figure 16. The peak of the droplet flux is shown to have shifted somewhat toward the larger droplet size in a 1-g environment. In general, in a 1-g environment, the droplets were larger and moved with slower axial velocity in comparison with the droplets studied in zero-gravity environment.

Maximum stable droplet size. - From the visual study of two-phase mercury vapor tow in a condenser tube (both in ground and airplane tests), it was observed that at a

TABLE VI. - ROOT-MEAN-SQUARE VALUES OF MEASURED DROPLET PARAMETERS IN 1-g ENVIRONMENT

Vapor	Droplet	Droplet	Maximum	Critical
quality,	height,	diameter,	stable	diameter,
x	δ _y ,	δ _Z ,	diameter,	$(\delta_{\mathbf{c}})_{\mathbf{x}},$
	in.	in.	$(\delta_{\max})_{x}$	in.
			in.	
0. 82	0.0110	0, 0134	0. 0182	0. 0051
. 36	. 0121	. 0170	. 0190	. 0063
. 20	. 0125	. 0178	. 0215	. 0093
. 03	. 0144	. 0187	. 0218	. 0120

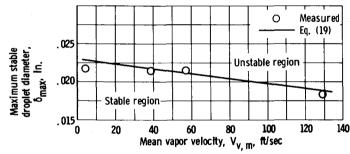


Figure 18. - Maximum stable droplet diameter as function of mean vapor velocity.

given vapor quality condition, certain droplets, which exceed a diameter of the most frequent droplet (see table VI), were moving stably close to a condenser wall with no oscillation due to turbulence. The droplets that obeyed the stability in flow and belonged to this size group were termed "maximum stable droplets". Therefore, in order to get a good statistical average diameter of a maximum stable droplet, 12 size measurements along the film were made of this category of droplets. The data obtained at various vapor velocities and quality conditions are tabulated in table VI, and the results are plotted as a function of a mean vapor velocity in figure 18. The straight-line equation

$$\delta_{\text{max}} = 0.0230 - 3.1 \times 10^{-5} \text{ V}_{\text{v,m}}$$
 (19)

fits the measured data within an accuracy of 2.5 rms percent. The

diameter of a maximum stable droplet size decreases with an increase in a mean vapor velocity. For example, at the vapor velocity of 4 feet per second, δ_{max} = 0.0218 inch, and at vapor velocity of 129 feet per second, δ_{max} = 0.0182 inch.

Equation (19) defines not the upper and not the lower limit of a stable droplet size, but it characterizes the statistical average obtained as a value of the root mean square. A knowledge of the maximum stable droplet size in a condenser tube is useful for several reasons. First, the maximum stable droplet size δ_{max} can itself be used in place of an average droplet size for making conservative estimates of mass transfer rates. Second, under some conditions, it may be possible to relate a given droplet size to maximum droplet size in order to predict tube slugging in a small diameter condenser tube.

Critical droplet diameter in ground tests. - The critical droplet diameter was measured from the droplet track left on a grayish condenser wall, such as shown in figure 19. Because of a small magnification factor of films ($\times 1.40$ to $\times 2.20$), the critical droplet diameter was not observed in the airplane tests. Measurement of the critical droplet

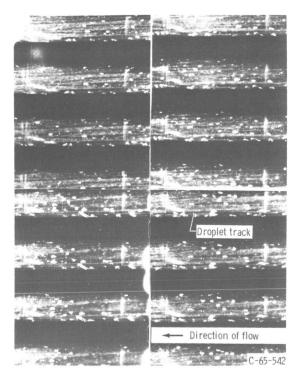


Figure 19. - Dropwise condensation in glass tube. Tracks left by moving droplets of variable width and equivalent to maximum droplet contact diameter. X8.9.

diameter at its incipient movement position was accomplished with the aid of crosshair micrometer dials with which the motion analyzer was equipped. For each vapor quality condition, from three to six such measurements were made along the film being analyzed. From these measurements, the average critical droplet diameter was determined as the root-meansquare value. The data obtained are listed in table VI, and the results are plotted as a function of vapor quality x in figure 20. As shown in the figure, the critical droplet diameter on a condenser wall is strongly dependent on the vapor phase quality. For example, for a vapor quality of 0.82, the critical droplet diameter is 0.0051 inch, and for a vapor quality of 0.03, the critical droplet diameter is 0.0120 inch.

The measured average values of the critical droplet diameter on a condenser wall were correlated with the theoretical values obtained by solving the force balance equation:

$$\delta_{\rm C} = \left(\frac{\mathscr{B} - \mathscr{C}}{\mathscr{A}}\right)^{3/2} \tag{20}$$

The derivation of this equation is given in appendix D.

The correlation of data, with the exception of droplets very close to the vapor-liquid interface, is fairly good. For example, at a vapor quality of 0.82, the correlation is within 3.0 percent, at 0.36, the correlation is within 4.0 percent, and at 0.20 and 0.03, the correlation deviates to approximately 15.0 percent. On the average, the theoretical equation (eq. (20)) correlates with the measured data points within 9 rms percent.

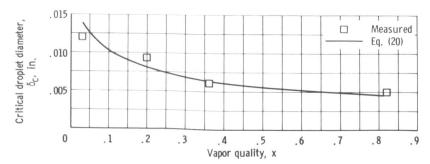


Figure 20. - Critical droplet diameter as function of vapor quality.

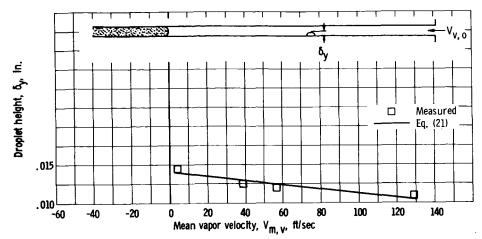


Figure 21. - Droplet height as function of vapor velocity.

Droplet height on condenser wall in ground tests. - In a study of frictional pressure drop of the vapor phase in turbulent flow through pipes, slowly moving droplets on a condenser wall may be regarded as the spherical segments that increase the effective roughness of the condenser tube wall. Therefore, the measured droplet height on a condenser wall may be regarded as an important roughness parameter δ_y/R , the so-called relative roughness introduced by Nikuradse in reference 8. In this case, the droplet height δ_y is regarded as a protrusion and R denotes the hydraulic radius of the cross section. From considerations of similtude it may be concluded that, in this case, the resistance coefficient depends on the relative roughness only, which is the determining factor of the frictional pressure drop in the boundary layer.

In the size range of the most frequent droplet, 12 measurements were made of the droplet height δ_y on a condenser wall. For a given vapor velocity and quality condition, the mean droplet height was taken as a value of the root mean square. The data, thus obtained, are tabulated in table VI, and the results are plotted in figure 21 as a function of mean vapor velocity. The straight line, expressed by the following relation

$$\delta_{y} = 0.0140 - 0.269 \times 10^{-4} V_{v, m}$$
 (21)

fits the plotted data points within an accuracy of 1.30 rms percent.

CONCLUDING REMARKS

The dominating factor in this analysis is the definition of the mercury droplet size distribution function in a uniformly cooled (q/A = constant), glass condenser tube in 1-g and zero-gravity environments. The observed droplet diameter and droplet frequencies followed the gamma distribution law, which expresses the fraction of droplets that have a

certain diameter. The proposed model of droplet size distribution at various vapor velocities and quality conditions along the condenser tube proved to be quite successful, and 1.72 rms percent agreement between the measured and calculated values was obtained. A correction factor was obtained to allow estimation of droplet distribution in a zerogravity environment from data obtained in a 1-g environment. It is anticipated that this analysis will furnish a convenient starting point for quantitative statistical description of droplet size distribution in a uniformly cooled condenser tube.

The critical droplet diameter on a condenser tube wall was defined by a theoretical expression derived from the force balance equation, which correlated with the measured values within an accuracy of 9 rms percent. Also, the measured values of the height of the most frequent droplet on a condenser wall and the diameter of the maximum stable droplet in turbulent vapor stream, at various vapor velocities and qualities, were defined by the best fit equations, which gave 1.30 and 2.50 rms percent deviation from the measured values, respectively. The maximum stable droplet size in a turbulent vapor flow, and also the height of a most frequent droplet on a condenser wall was expressed as a function of a mean vapor velocity.

It was also observed that for a given flow condition, the droplets were more uniformly dispersed throughout the vapor phase and on the condenser tube wall in a zero-gravity environment than in a 1-g environment. In addition to this, droplets moved with a more uniform axial velocity (less turbulent) in zero gravity than in a 1-g environment.

Lewis Research Center,

National Aeronautics and Space Administration, Cleveland, Ohio, September 28, 1966, 701-04-00-02-22.

APPENDIX A

SYMBOLS

A	condenser surface area	$n_{\mathbf{v}}$	average number of droplets per
A, A, C, D	ratios of parameters necessary in describing droplet size		unit volume of a condenser tube
	distribution law for zero-	n _{w,i}	percentage of droplets on wall
	gravity to that of 1-g environ-	P	pressure
- h	ment	q	heat flux, Btu/hr
a,b, c,d	parameters necessary to de- scribe gamma law of droplet size	Re	droplet Reynolds number, ${{ m G}}{^{ m C}}_{ m c}/{^{\mu}}_{ m V}$
f	average number of droplets be-	T	temperature
	tween two length marks	v	velocity
$\mathbf{f_i}$	number of droplets found in i th	v	volume
	group interval, absolute	$\mathbf{w}_{ ext{tot}}$	total weight rate flow
C	group-mark frequency mass velocity, w/A	x	vapor quality, \dot{w}_v/\dot{w}_{tot}
G	•,	x, y, z	Cartesian coordinates
h	rectangle height	β	ratio of w _v w ₁
L l	condenser tube length distance between two length	Δ	ratio of $\left(\delta_{\mathrm{mf}}\right)_{0} / \left(\delta_{\mathrm{mf}}\right)_{1}$
	marks on a condenser tube	Δδ	rectangle width of histogram on
m	number of intervals		δ-axis
N	total (average) number of drop-	$^{\delta}{}_{\mathbf{b}}$	laminar-sublayer thickness
	lets between two length marks	$^{\delta}_{ m c}$	critical droplet diameter
$N(\delta)$	droplet size distribution func-	$\delta_{\mathbf{i}}$	variable droplet diameter
	tion	$\delta_{\mathbf{y}}$	measured droplet height
n	droplet frequency	$oldsymbol{\delta}_{\mathbf{z}}$	droplet diameter
ⁿ s,i	percentage of droplets in	μ	viscosity
n	stream of vapor core	ρ	density
n _t	average number of droplets per unit time crossing any refer- ence point along condenser	σ	surface tension, standard de- viation
	tube	τ	time

Δau	time interval for droplet to travel	l	liquid
	between two length marks,	s	stream of vapor core
	Number of frames recorded , sec Frame rate	sat	saturation
φ	droplet flux, average number of	sh	superheat
,	droplets crossing per unit area	t	time
	per unit time	tot	total
Subsc	ripts:	v	vapor, volume
act	actual	w	wall
c	critical	у	height
i	i th interval	Z	diameter
m	mean	δ	droplet
mf	most frequent	0	zero-gravity environment
max	maximum	1	1-g environment

APPENDIX B

DROPLET SIZE DISTRIBUTION FUNCTION

From the measured data, both the droplet diameters $\,\delta\,$ and droplet frequencies $\,$ n followed the gamma distribution law. The gamma probability density function is given in reference 6 by

$$f(\delta) = \frac{\lambda}{\Gamma(r)} (\lambda \delta)^{r-1} e^{-\lambda \delta} \qquad \delta > 0$$
 (B1)

where r and λ are positive constants.

An area of the rectangle of the histogram which represents the relative frequency is given by $n \Delta \delta$. Therefore, multiplying equation (B1) by this factor results in

$$f(\delta)n \ \Delta \delta = n \ \Delta \delta \ \frac{\lambda}{\Gamma(r)} (\lambda \delta)^{r-1} e^{-\lambda \delta}$$

Letting

$$N(\delta) = f(\delta)n \Delta \delta$$

$$c = n \, \Delta \delta \, \frac{\lambda}{\Gamma(r)}$$

$$a = \lambda$$

and

$$b = (r - 1)$$

gives equation (B1) in the form

$$N(\delta) = c(a\delta)^{b} e^{-a\delta}$$
 (B2)

where a, b, and c are the three characteristic parameters that make a frequency function $N(\delta)$ to be determined from the measured data.

The relation among the three characteristic parameters in equation (B2) was determined from the following two imposed conditions:

(1) By definition, the total area under the distribution curve defining the frequency function $N(\delta)$ is equal to 1. Thus,

$$\int_0^\infty N(\delta)d\delta = 1 \tag{B3}$$

Even though the droplet size has been measured from $\delta=0$ to $\delta=\delta_{max}$, the largest droplet size found, infinity as the upper limit of the integral was used in equation (B3) in order to simplify the calculations. The use of this limit does not introduce any appreciable error, since no excessive large droplets exist, and the distribution function $N(\delta)$ must approach zero very rapidly for $\delta>\delta_{max}$.

Substituting equation (B2) into equation (B3) yields

$$\int_0^\infty c(a\delta)e^{-a\delta} d\delta = 1$$
 (B4)

Making the substitution $x = a\delta$ into equation (B4) gives

$$\int_0^\infty x^b e^{-x} dx = \frac{a}{c}$$
 (B5)

The integral may be evaluated by the gamma function thus giving the relation among the three parameters from the first imposed condition

$$\frac{a}{c} = \Gamma(b+1) \tag{B6}$$

(2) The gamma distribution curve exhibits a maximum that may be located by differentiating equation (B2) with respect to δ and setting the result equal to zero. Thus,

$$\frac{d}{d\delta}N(\delta) = 0 = -ac(a\delta)^b e^{-a\delta} + abc(a\delta)^{b-1} e^{-a\delta}$$

Solving for δ_{mf} gives

$$\delta_{\rm mf} = \frac{\rm b}{\rm a} \tag{B7}$$

The relation of the third parameter of the normal distribution function was obtained by substituting equation (B7) into equation (B6) and eliminating the parameter a. Thus,

$$c = \frac{b}{\delta_{mf}} \frac{1}{\Gamma(b+1)}$$
 (B8)

The droplet diameter δ_{mf} is determined from the gamma distribution function, the frequency of occurrence of which is the maximum. The experimental value of δ_{mf} is known.

Substitution of equations (B7) and (B8) into equation (B2) eliminates the parameters a and c, which gives the maximum value $N(\delta_{mf})$ of the gamma distribution function. Thus,

$$N(\delta_{mf})\delta_{mf} = \frac{b^{b+1}}{\Gamma(b+1)} e^{-b}$$
(B9)

Since equation (B9) is a transcendental function in b, $N(\delta_{mf})\delta_{mf}$ was plotted as a function of b, and for each value of $N(\delta_{mf})\delta_{mf}$ obtained from the measurements, the parameter b was obtained from this plot. The numerical values of the parameters a and c were calculated from equations (B7) and (B8), respectively.

APPENDIX C

VAPOR VELOCITY DISTRIBUTION ALONG CONDENSER TUBE

If q/A is constant, as it will be for condensing vapors, it is reasonable to assume that a mean vapor velocity will have a linear distribution along the condenser length. Thus,

$$V_{v,m} = V_{v,o} \left(1 - \frac{z}{L} \right) \tag{C1}$$

Consider the condenser tube at the region of inlet, as shown in figure 22, where P_4 is the measured pressure, equal to 18.30 pounds per square inch absolute, P_6 is the inlet pressure, equal to 0.90 P_4 , T_{sat} = 685° F or 1145° R, which corresponds to P_6 , T_{act} = 875° F or 1335° R measured temperature, the vapor quality is 90 percent, ρ_{sat} is 0.270 pound per cubic foot, which corresponds to P_6 , and ρ_{sat} = 0.270 $\frac{1145}{1335}$ = 0.232 pound per cubic foot.

For a constant diameter tube, the vapor velocity may be defined by

$$V_{v} = \frac{\dot{x}w}{\rho_{v}A} \tag{C2}$$

Thus, the vapor velocity corresponding to saturation conditions will be

$$\left(V_{v}\right)_{sat} = \frac{0.090(0.0385)(144)}{0.270\frac{\pi}{4}(0.440)^{2}} = \frac{5.000}{0.0412} = 121 \text{ ft/sec}$$

and the inlet vapor velocity corrected due to superheat results in

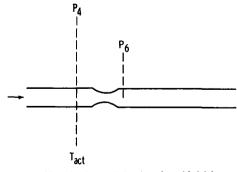


Figure 22. - Condenser tube at region of inlet from boiler.

$$(V_{v,o})_{sh} = \frac{0.270}{0.232}$$
 121 = 141 ft/sec

Substituting this value of $(V_{v,o})_{sh}$ into equation (C1) results in

$$V_{v, m} = 141 \left(1 - \frac{z}{L}\right)$$
 (C3)

APPENDIX D

CRITICAL DROPLET SIZE

Consider a microscopic droplet resting on a solid surface under the influence of static forces, gravity, and drag, as illustrated in figure 23. The retarding force that re-

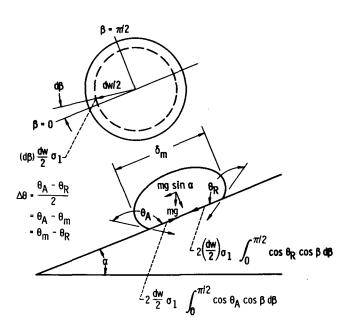


Figure 23. - Vector analysis of forces acting on droplet at moment of incipient movement,

sists droplet movement is attributed to adhesion force (attraction force exerted between the molecules of liquid droplet and the molecules of solid surface). It is assumed that, for a small droplet, adhesion tension forces at the juncture of the liquid-solid surface will keep the droplet stationary on the tube wall in spite of the drag and gravity forces on the droplet. If droplet size increases above a critical size, the drag and gravity forces will exceed the resisting adhesion tension forces at the wall, and the droplet will begin to move. The shape of a stationary droplet is assumed to be unaffected by the drag or gravity forces. Thus,

$$\sum \mathbf{F} = \mathbf{0}$$

Adhesion force ± Gravity force - Drag force = 0

Changing the signs in the preceding equation gives

-Adhesion force ± Gravity force + Drag force = 0

From figure 23, the equation of motion is

$$-2K_{1} \frac{d_{\mathbf{w}}}{2} \sigma_{l} \left(\cos \theta_{\mathbf{R}} + \cos \theta_{\mathbf{A}}\right) \int_{0}^{\pi/2} \cos \beta \, d\beta \pm K_{2} v_{\delta} \frac{\rho_{l}}{g_{c}} g \sin \alpha + K_{3} A_{\delta} \frac{\rho_{\mathbf{v}} V_{\mathbf{v}}}{2g_{c}} C_{\mathbf{D}} = 0$$
(D1)

where

g acceleration due to gravity

g conversion factor

C_D drag coefficient on droplet

α surface inclination relative to horizontal

Substituting the following trigonometric expressions

$$\theta_{\mathbf{A}} = \theta_{\mathbf{m}} + \Delta \theta$$

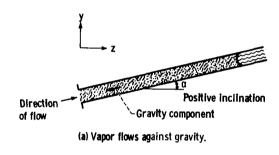
$$\theta_{\mathbf{R}} = \theta_{\mathbf{m}} - \Delta \theta$$

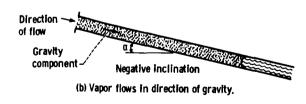
and

$$\cos \theta_{\mathbf{R}} + \cos \theta_{\mathbf{A}} = 2 \cos \theta_{\mathbf{m}} \cos \Delta \theta$$

into equation (D1) and integrating result in

$$-2d_{\mathbf{w}}\sigma_{l}\cos\theta_{\mathbf{m}}\cos\Delta\theta \pm K_{2}V_{\delta}\frac{\rho_{l}}{g_{l}}g\sin\alpha + K_{3}A_{\delta}\frac{\rho_{\mathbf{v}}V_{\mathbf{v}}}{2g_{l}}C_{\mathbf{D}} = 0$$
 (D2)







(c) Gravity has negligible effect when $\alpha \to 0$ or $g/g_C \to 0$. Figure 24. - Three extreme conditions of flow in condenser tube. Equation (D2), with respective signs, applies for all three cases shown in figure 24. In this analysis, for a horizontal tube when $\alpha = 0$, the gravity term drops out, and equation (D2) reduces to the following form:

$$-2K_1\sigma_l \cos \theta_m \cos \Delta\theta$$

$$+ K_3 A_\delta \frac{\rho_v V_v}{2g_c} C_D = 0$$
 (D3)

The quantities K_1 , K_2 , and K_3 in equation (D2) are the shape factors of a droplet distorted by the curvature of the tube wall defined as follows:

Perimeter shape factor of a droplet contact with tube wall:

$$K_1 = \frac{\text{Measured perimeter of droplet contact with tube wall}}{\text{Apparent perimeter of droplet contact with tube wall}}$$

Volume shape factor:

$$K_2 = \frac{\text{Volume of spherical segment having contact angle of } 110^{\text{O}} \text{ approximately}}{\text{Volume of equivalent spherical droplet}}$$

Cross-sectional area shape factor:

$$K_3 = \frac{\text{Cross-sectional area of droplet on glass plate}}{\text{Cross-sectional area of equivalent droplet in glass tube}}$$

In order to obtain statistical averages for K_1 , K_2 , and K_3 , 12 mercury droplets ranging in diameter from 0.0051 to 0.0120 inch were photographed on a glass plate and in a glass tube. From enlarged images of a droplet (side view, bottom view, and top view), the shape factors were determined statistically from the measured droplet dimensions.

Using the following numerical values of the shape factors,

$$K_1 = 1.23$$

$$K_2 = 0.832$$

$$K_3 = 0.875$$

and the following properties of mercury at the temperature of saturation of 680° F, from reference 9,

$$\rho_l = 794 \text{ lb/cu ft}$$

$$\rho_{\rm vr} = 0.260$$
 lb/cu ft

$$\mu_{\rm v}$$
 = 0.148 lb/ft hr

$$\sigma_l = 0.027 \text{ lb/hr}$$

and

$$V_{v}^{2} = \frac{G^{2}}{\rho_{v}^{2}} x^{2}$$

where $G = w/A = 131\ 300$ pounds per square foot per hour, and d_w is approximately equal to δ_m gives equation (D3) in the form

$$-19.880 \, \delta_{\rm c}^2 \, \frac{{\rm G}^2}{{\rm g}_{\rm c}} \, {\rm x}^2 {\rm C}_{\rm D} + \delta_{\rm l} \, \cos \, \theta_{\rm m} \, \cos \, \Delta \theta = 0 \tag{D4}$$

Evaluation of drag coefficient. - In the critical droplet size range investigated (0.0051 $\leq \delta_{m} \leq$ 0.0120 in.) it may be safe to assume that the critical droplet formed on a condenser wall will be within the thickness of a laminar sublayer. This assumption is based on the laminar sublayer theory in turbulent flow developed by Eckert in reference 4.

A well-known expression for the drag coefficient $\,C_D^{\,}$ is given in reference 10. In the laminar sublayer, the drag coefficient may be expressed by

$$C_{D} = \frac{24}{Re} + \frac{4.8}{Re^{1/3}}$$

Defining the Reynolds number of a critical droplet size by

$$Re = \frac{G\delta_{c}x}{\mu_{v}}$$

gives the drag coefficient:

$$C_{D} = \frac{24\mu_{v}}{G\delta_{c}x} + \frac{4.8 \ \mu_{v}^{1/3}}{G^{1/3} \ \delta_{c}^{1/3} \ x^{1/3}}$$

Substituting this expression for C_D , the numerical value of the conversion factor

$$g_c = 4.17 \times 10^8 \frac{\text{(lb mass)(ft)}}{\text{(lb force)(hr}^2)}$$

and

$$\mu_{\rm v} = 0.148 \, \frac{\rm lb \, mass}{\rm (ft)(hr)}$$

into equation (D4) and simplifying result in

$$\left(0.945\times10^{-8}G^{5/3}x^{5/3}\delta_{c}^{5/3}+16.934\times10^{-8}Gx-\cos\theta_{m}\Delta\theta\right)\delta_{c}=0$$
 (D4)

Equation (D4) is the force-balance equation of a mercury droplet at its incipient movement in a horizontal glass tube. Solving equation (D4) for the droplet diameter gives an expression for the critical droplet size δ_c

$$\delta_{\rm c} = \left(\frac{\mathscr{Z} - \mathscr{C}}{\mathscr{A}}\right)^{3/2} \tag{D5}$$

where

$$\mathscr{A} = 0.945 \times 10^{-8} G^{5/3} x^{5/3}$$
$$\mathscr{A} = \cos \theta_{m} \cos \Delta \theta$$
$$\mathscr{C} = 16.934 \times 10^{-8} Gx$$

The following inputs obtained from the measurements

$$\theta_{\rm m} = 110^{\rm O}$$

$$\Delta \theta = 10^{\rm O}$$

G = 131 300 lb/(sq ft)(hr)

were used to solve equation (D5) in the range of vapor quality analyzed (0.010 \leq x \leq 0.900) by the use of a digital computer.

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